

PhD presentation on

“Size Effects in Semi-brittle Materials and Gradient Theories with Application to Concrete”

by

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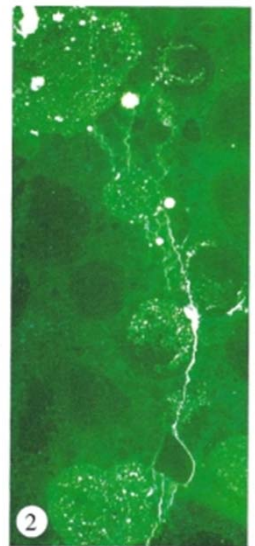
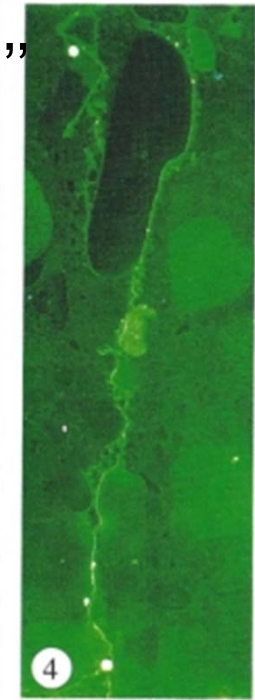
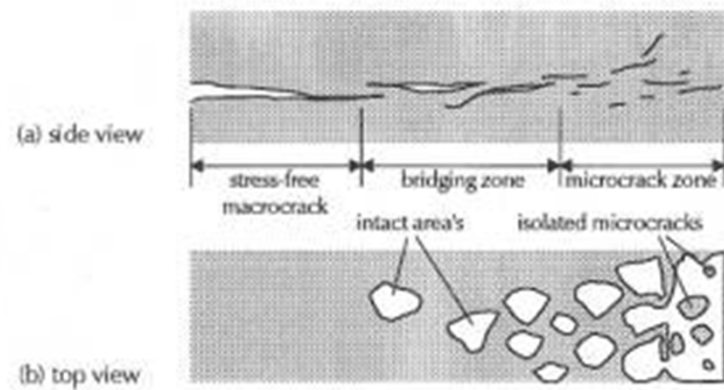
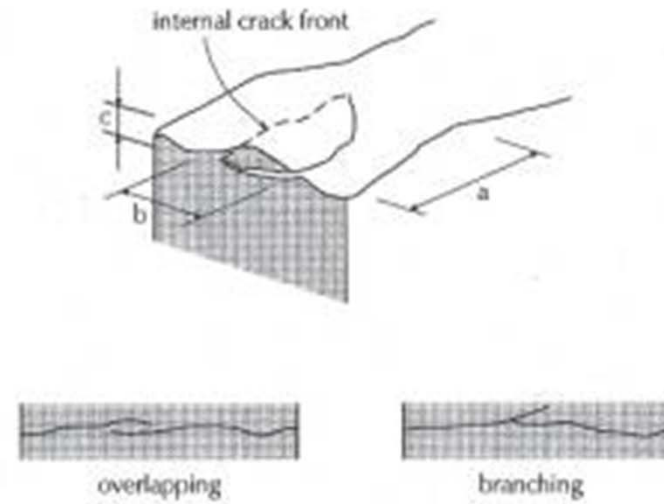
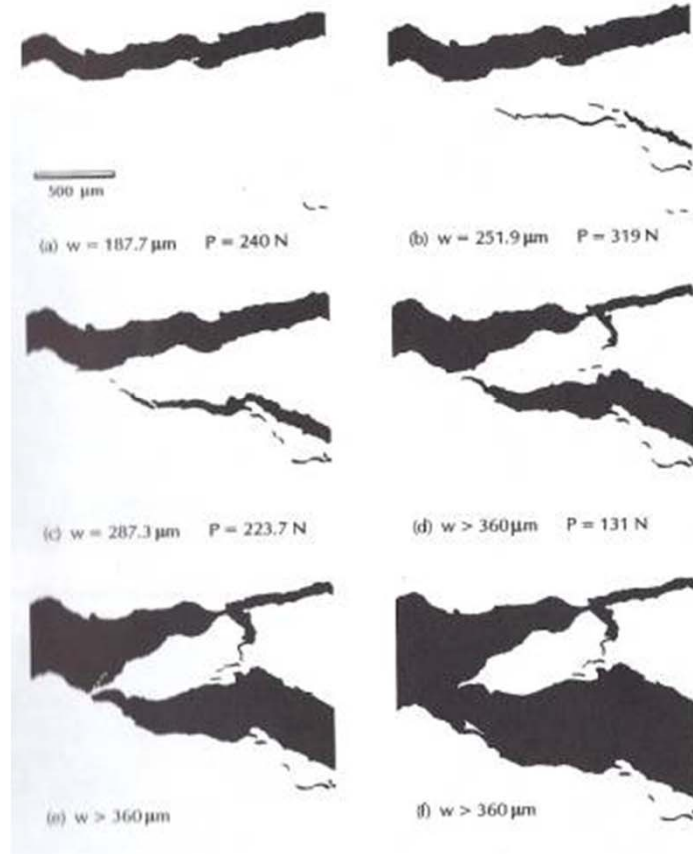
Advisor: Prof. Philip C. Perdikaris

Co-advisor: Prof. Antonios E. Giannakopoulos

January, 2015

INTRODUCTION

Microcracking: "branching" and "crack face bridging"

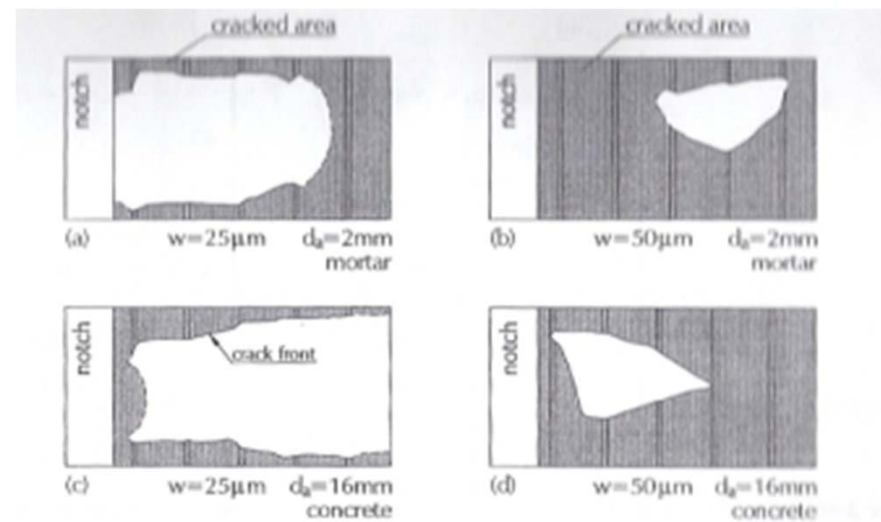
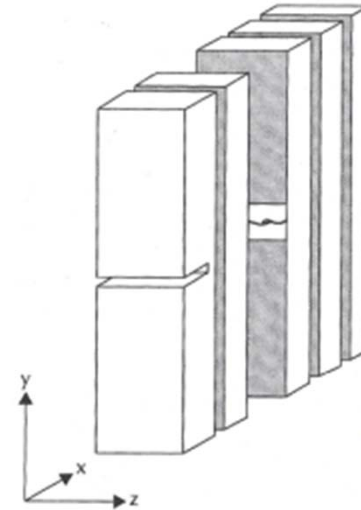
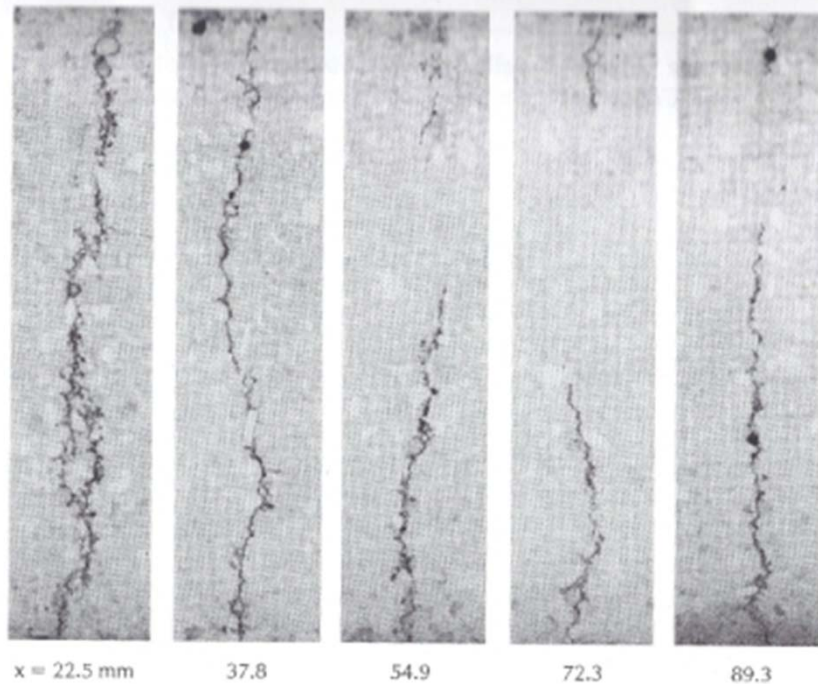


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INTRODUCTION

Complexity of the observed behavior of microcracking

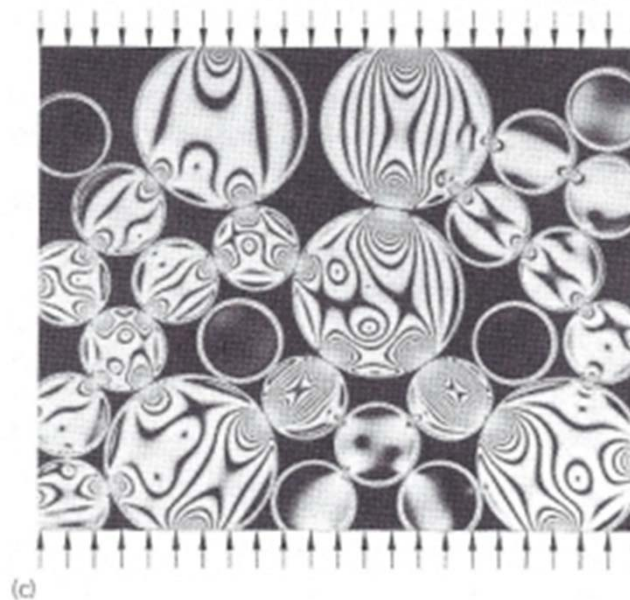
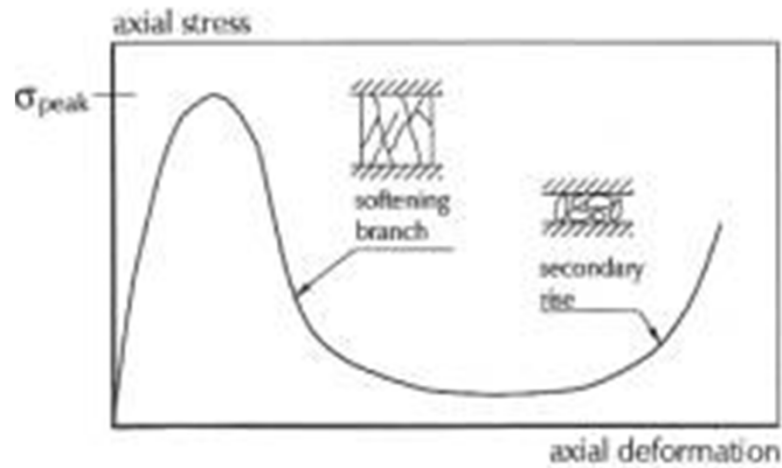
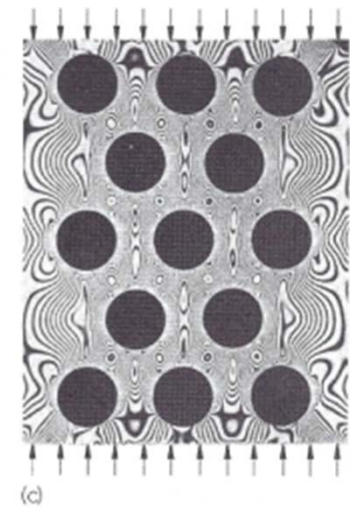
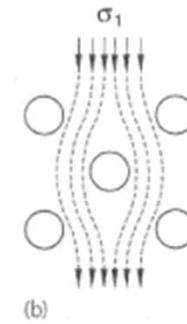
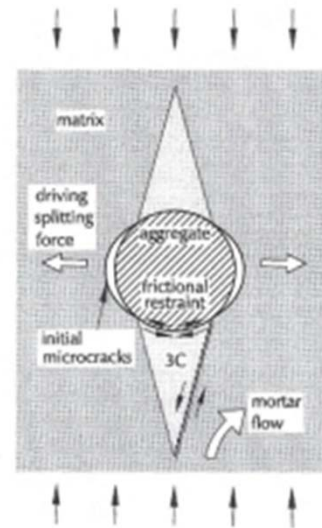
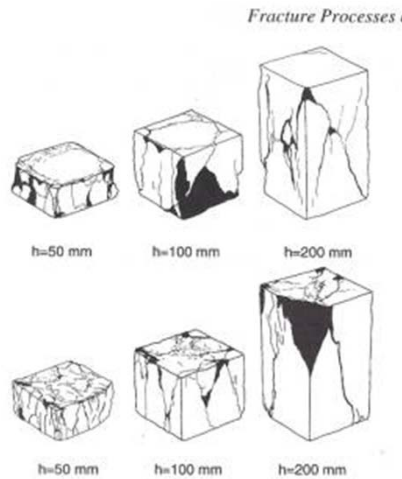
Mode I fracture



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INTRODUCTION

Mode I cracking in compression

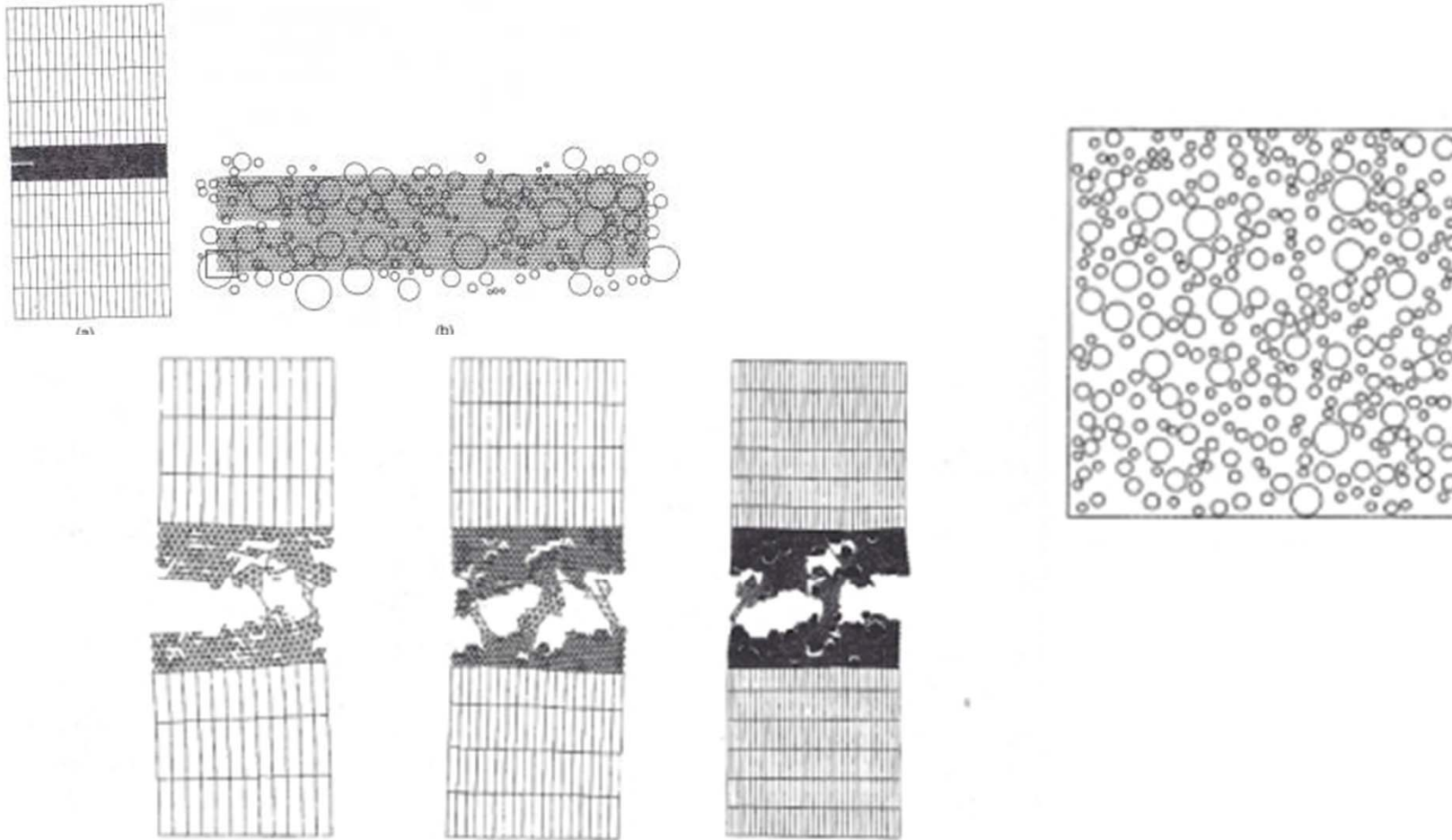


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INTRODUCTION

Lattice models: Modeling the full details of the microstructure

most important: particle structure



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INTRODUCTION

Ability to predict the complex behavior of microcracking

Fracture Mechanics for Structural Analysis

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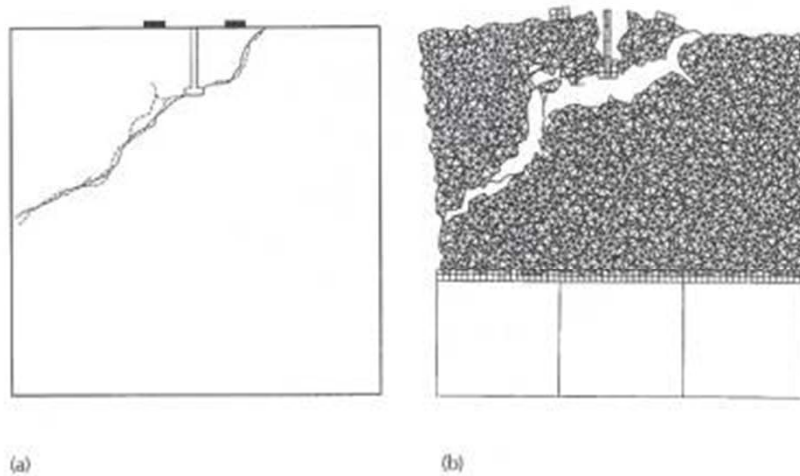
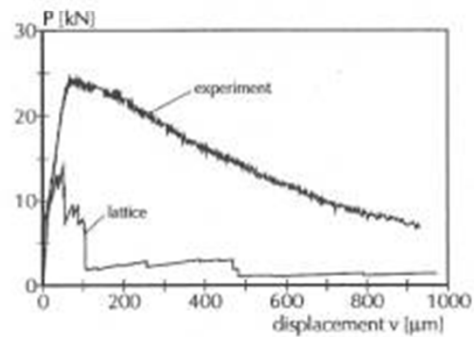


Figure 8.13 Crack growth observed in experiments with $a = d/2$ (a), and from lattice analyses (b).⁵⁰¹



Fracture Mechanics for Structural Analysis

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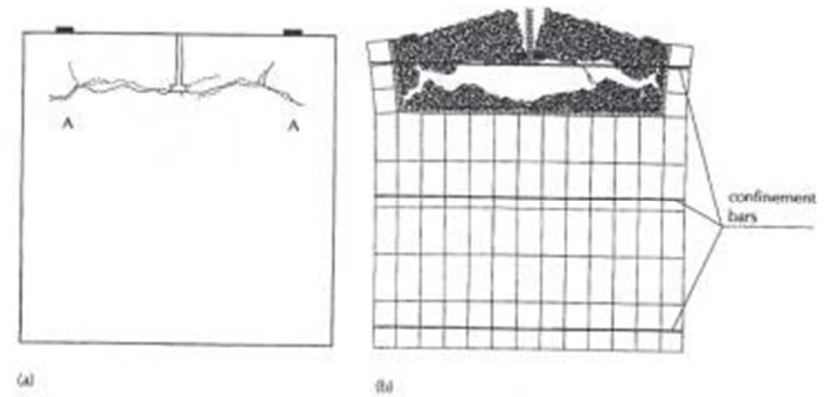
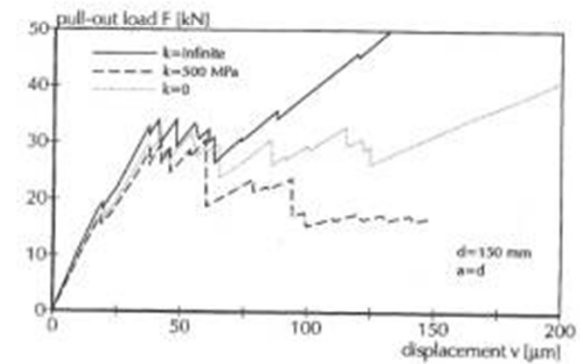


Figure 8.16 Experiment (a) and numerical lattice simulation of a laterally confined specimen with a support span $a = 2d$ ($d = 150$ mm).⁵⁰²



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INTRODUCTION

Dipolar elasticity

Strain gradient elasticity: $\sigma_{xx} = (1 - g^2 \nabla^2) E \varepsilon_{xx}$ (1D, $\nu=0$)

Hooke's law extended to the 2nd spatial derivative of the Cauchy strain

Non-local models:

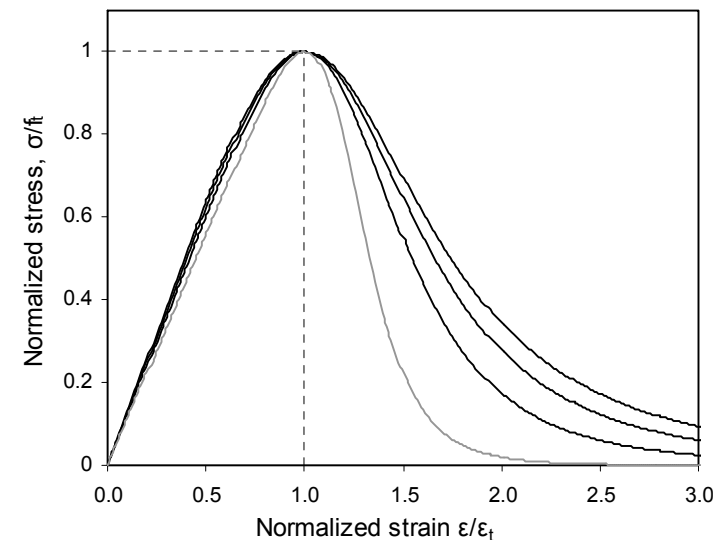
A length scale parameter is needed for modeling materials such as concrete exhibiting size effect

Stress-strain softening law:
$$\frac{\sigma}{f_i} = \frac{\beta_i (\varepsilon / \varepsilon_i)}{\beta_i - 1 + (\varepsilon / \varepsilon_i)^{\beta_i}}$$

Size effect in elasticity:

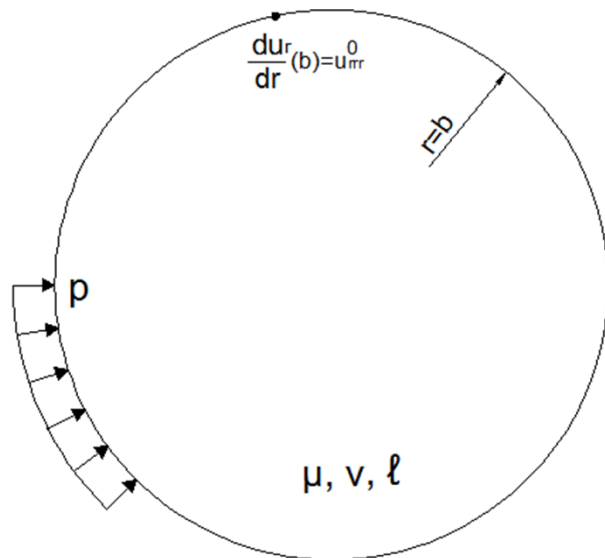
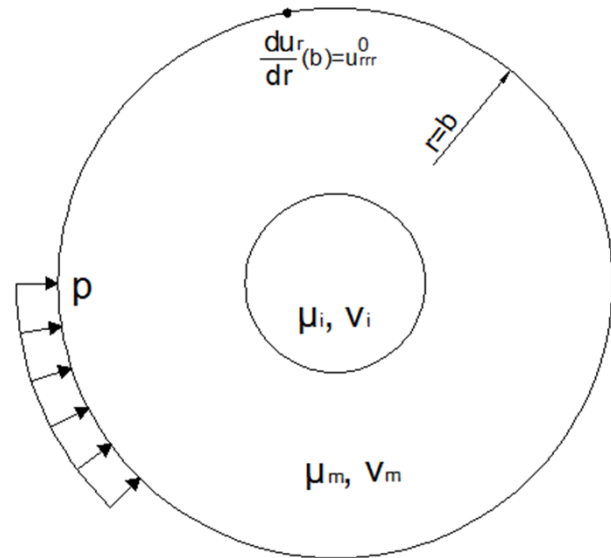
Stiffer static response than that predicted by classical elasticity if $g > 0$

What is a physical interpretation of the internal length?



HOMOGENIZATION ESTIMATES OF THE INTERNAL LENGTH

The simplest case of composite material (2D case of circular inclusions)



Input parameters

Matrix and inclusion elasticity properties

Composition value, $c=a/b$

Effective material properties of composite material

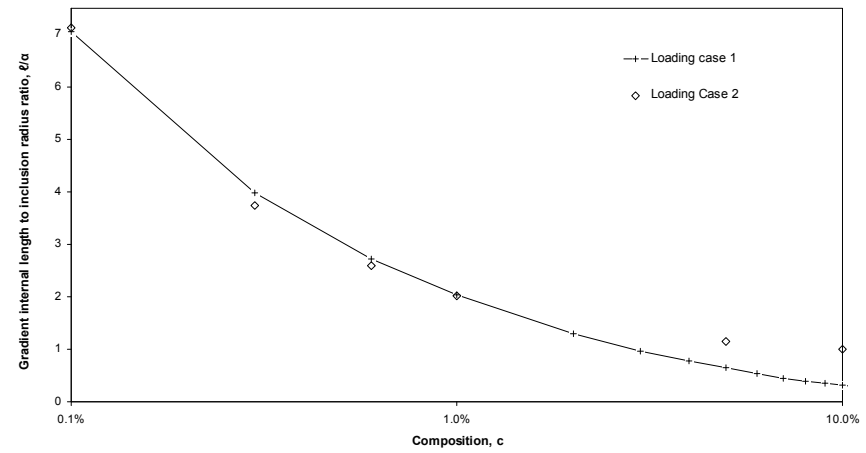
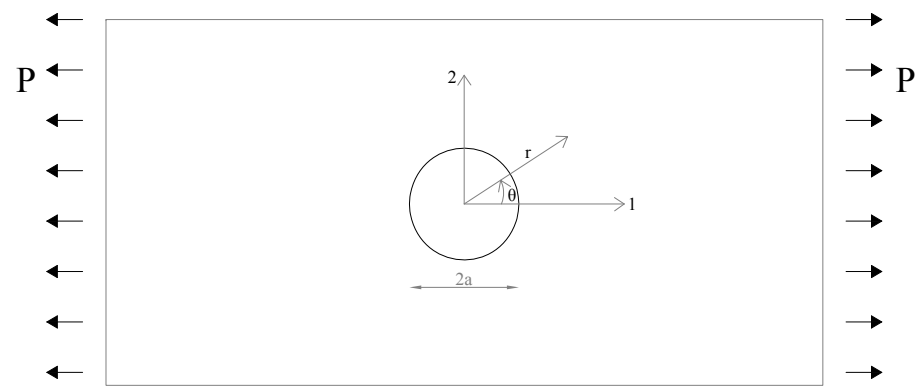
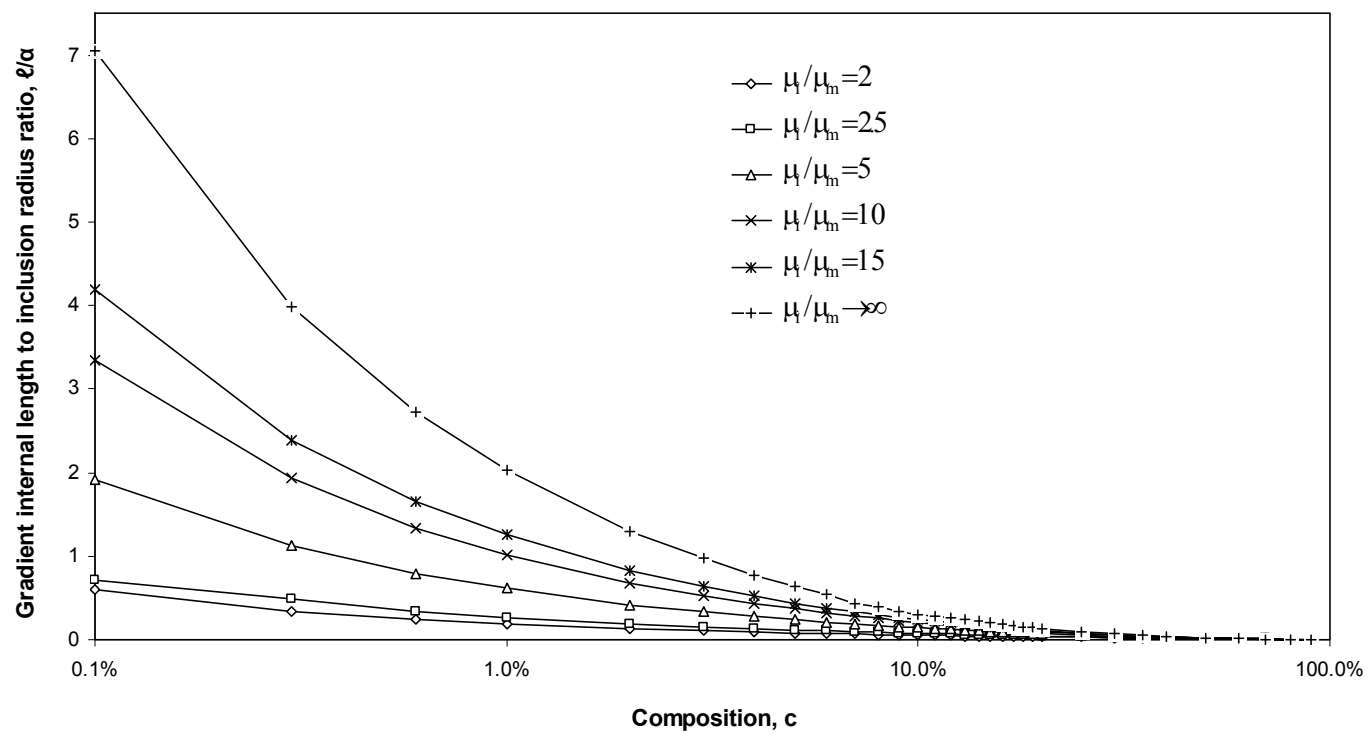
$$\mu = f_1(\mu_m, \nu_m, \mu_i, \nu_i, c)$$

$$\nu = f_2(\mu_m, \nu_m, \mu_i, \nu_i, c)$$

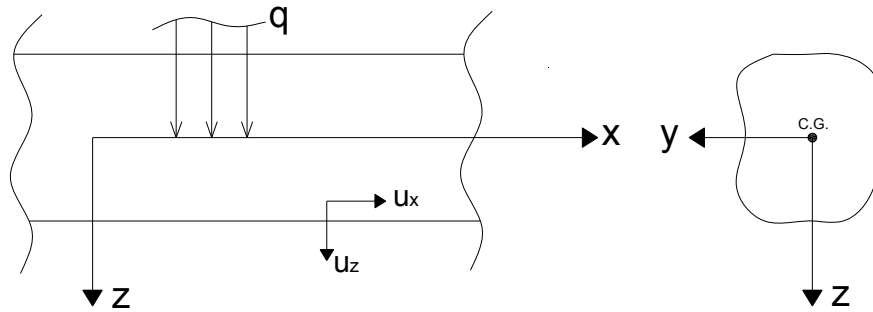
Estimation of the internal length

$$U_{cl} = U_{gr}$$

HOMOGENIZATION ESTIMATES OF THE INTERNAL LENGTH



STRUCTURAL ANALYSIS USING A DIPOLAR ELASTIC TIMOSHENKO BEAM



Timoshenko kinematics

$$\begin{aligned} u_x &= z \psi(x) \\ u_y &= 0 \\ u_z &= w(x) \end{aligned}$$

Cauchy stresses

$$\begin{aligned} \bar{\sigma}_{xx} &= E \varepsilon_{xx} \\ \bar{\sigma}_{xz} &= kG \gamma_{xz} \end{aligned}$$

Total strain energy:

$$U_{\text{tot}} = U_{\text{cl}} + U_{\text{gr}}$$

$$U_{\text{cl}} = \frac{1}{2} \iiint_V \left(\bar{\sigma}_{xx} \varepsilon_{xx} + 2 \bar{\sigma}_{xz} \varepsilon_{xz} \right) dx dy dz$$

$$U_{\text{gr}} = \frac{1}{2} g^2 \iiint_V \left(\frac{\partial \bar{\sigma}_{xx}}{\partial x} \frac{\partial \varepsilon_{xx}}{\partial x} + 2 \frac{\partial \bar{\sigma}_{xz}}{\partial x} \frac{\partial \varepsilon_{xz}}{\partial x} + \underbrace{\frac{\partial \bar{\sigma}_{xx}}{\partial z} \frac{\partial \varepsilon_{xx}}{\partial z}} \right) dx dy dz$$

Term that should not be omitted

Non-zero strains:

$$\varepsilon_{xx} = \frac{\partial u_x}{\partial x} = z \frac{d\psi}{dx}$$

$$\gamma_{xz} = 2\varepsilon_{xz} = \frac{\partial u_x}{\partial z} + \frac{\partial u_z}{\partial x} = \frac{dw}{dx} + \psi$$

STRUCTURAL ANALYSIS USING A DIPOLAR ELASTIC TIMOSHENKO BEAM

$$\delta U_{\text{tot}} = \int_0^L \left\{ \begin{aligned} &\delta\psi \left(-EI \left(1 - g^2 \frac{d^2}{dx^2} \right) \frac{d^2\psi}{dx^2} + kAG \left(1 - g^2 \frac{d^2}{dx^2} \right) \left(\frac{dw}{dx} + \psi \right) - EA g^2 \frac{d^2\psi}{dx^2} \right) \\ &+ \delta w \left(-kAG \left(1 - g^2 \frac{d^2}{dx^2} \right) \left(\frac{d^2w}{dx^2} + \frac{d\psi}{dx} \right) \right) \end{aligned} \right\} dx$$

$$+ \left[\delta\psi \left(EI \left(1 - g^2 \frac{d^2}{dx^2} \right) \frac{d\psi}{dx} + kAG g^2 \left(\frac{d^2w}{dx^2} + \frac{d\psi}{dx} \right) + EA g^2 \frac{d\psi}{dx} \right) \right]_0^L$$

$$+ \left[\delta w \left(kAG \left(1 - g^2 \frac{d^2}{dx^2} \right) \left(\frac{dw}{dx} + \psi \right) \right) \right]_0^L$$

$$+ \left[\delta\psi' \left(g^2 EI \frac{d^2\psi}{dx^2} \right) \right]_0^L$$

$$+ \left[\delta w' \left(kAG g^2 \left(\frac{d^2w}{dx^2} + \frac{d\psi}{dx} \right) \right) \right]_0^L$$

4 independent BC

w, w', ψ, ψ'

(bold are the classical elasticity variables)

4 energetically conjugate quantities

Q, Y, M, m

(Y, m: double shear or bi-force and double moment or bi-moment)

$$\delta W = \int_0^L q \delta w dx + [Q \delta w]_0^L + [M \delta \psi]_0^L + [Y \delta w']_0^L + [m \delta \psi']_0^L$$

STRUCTURAL ANALYSIS USING A DIPOLAR ELASTIC TIMOSHENKO BEAM

Equilibrium equations

$$\left(1 + \frac{A}{I}g^2 - g^2 \frac{d^2}{dx^2}\right) \frac{d\bar{M}}{dx} = \left(1 - g^2 \frac{d^2}{dx^2}\right) \bar{Q} \quad \left(1 - g^2 \frac{d^2}{dx^2}\right) \frac{d\bar{Q}}{dx} = -q$$

Boundary conditions

$$\left[\left\{ Y - \left(g^2 \frac{d\bar{Q}}{dx} \right) \right\} \delta w' \right]_0^L = 0 \quad \left[\left\{ Q - \left(\left(1 - g^2 \frac{d^2}{dx^2} \right) \bar{Q} \right) \right\} \delta w \right]_0^L = 0$$

$$\left[\left\{ M - \left(\left(1 - g^2 \frac{d^2}{dx^2} \right) \bar{M} + \frac{A}{I} g^2 \bar{M} + g^2 \frac{d\bar{Q}}{dx} \right) \right\} \delta \psi \right]_0^L = 0 \quad \left[\left\{ m - \left(g^2 \frac{d\bar{M}}{dx} \right) \right\} \delta \psi' \right]_0^L = 0$$

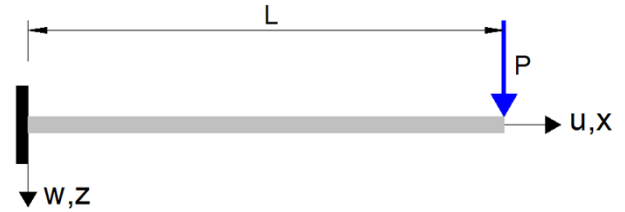
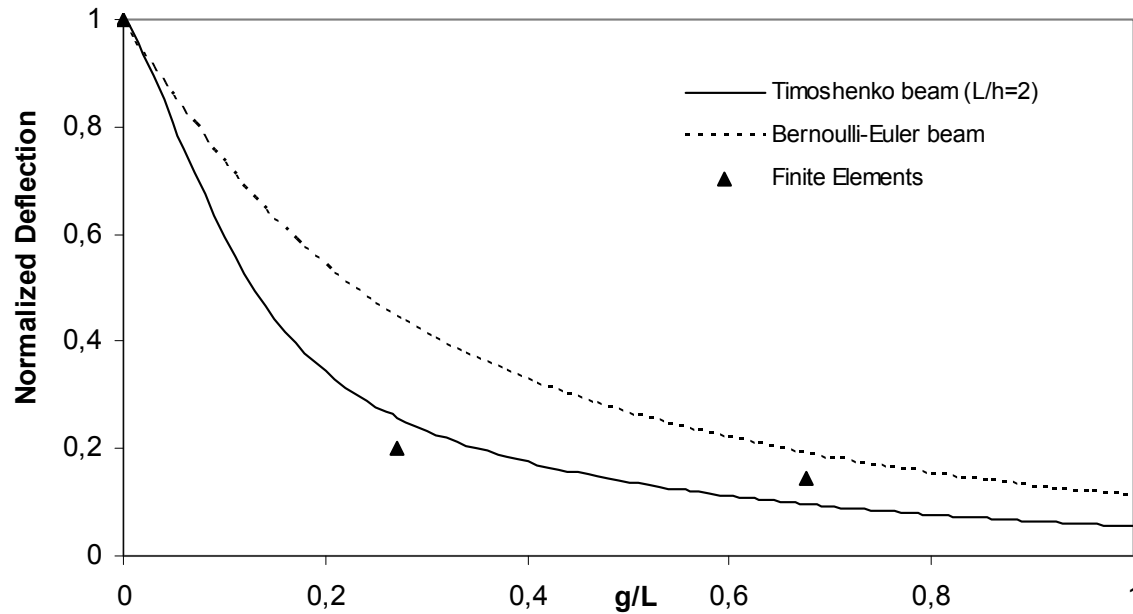
Closed-form solution

$$w(x) = \left\{ \begin{aligned} & (c_3 - d_3)x - \frac{d_4}{2}x^2 + c_4 + d_2\ell^3 e^{-x/\ell} - d_1\ell^3 e^{x/\ell} + c_1g^2 e^{x/g} + c_2g^2 e^{-x/g} \\ & + \frac{q}{24EI} \left(\frac{\ell}{g} \right)^2 x^4 - \frac{q}{2kAG} x^2 - \frac{c_3 kAG}{6EI} \left(\frac{\ell}{g} \right)^2 x^3 \end{aligned} \right\}$$

$$\psi(x) = -\frac{q}{6EI} \left(\frac{\ell}{g} \right)^2 x^3 + \frac{kAG}{2EI} \left(\frac{\ell}{g} \right)^2 c_3 x^2 + d_1 \ell^2 e^{x/\ell} + d_2 \ell^2 e^{-x/\ell} + d_3 + d_4 x$$

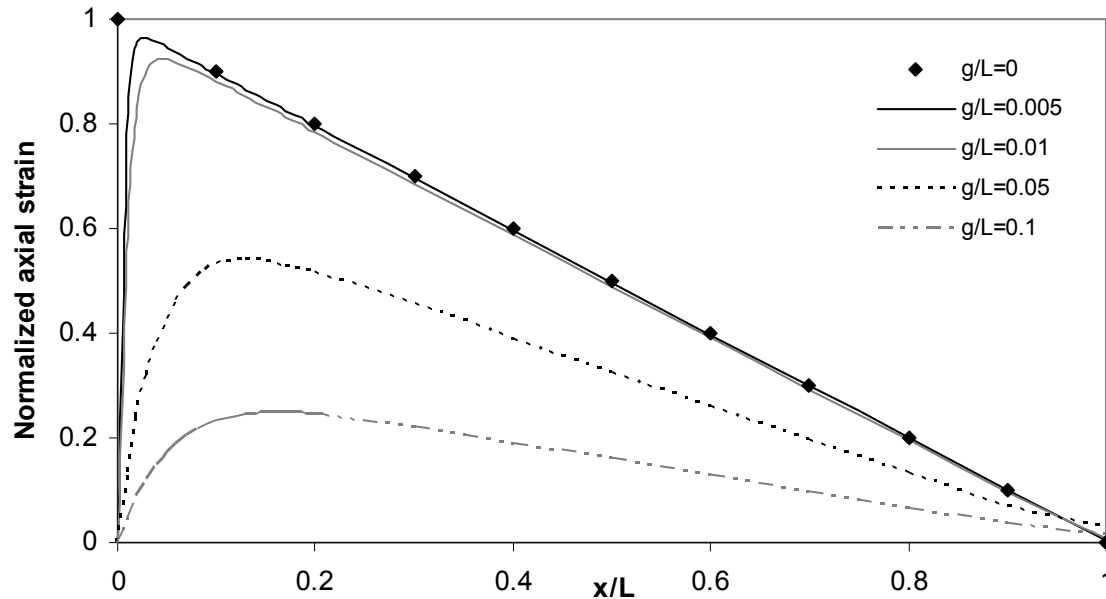
8 constants to be determined from 8 B.C.'s (4 at each end)

STRUCTURAL ANALYSIS USING A DIPOLAR ELASTIC TIMOSHENKO BEAM



Comparison with FEM results

+ Effect of g/h



Clamped end: Boundary layer effect as in BE beam

Free end: zero axial strains

NON-LOCAL DAMAGE MODEL

Gibb's energy: $G = \frac{1}{2} \boldsymbol{\tau} : \mathbf{C} : \boldsymbol{\tau} + \frac{1}{2} \boldsymbol{\lambda} \cdot \mathbf{B} : \boldsymbol{\lambda} - A^c$ $\boldsymbol{\lambda} = g^2 \nabla \boldsymbol{\tau}$ $\mathbf{B} = (1/g^2) \mathbf{C}$

Microcracking adds flexibility: $\mathbf{C} = \mathbf{C}^0 + \mathbf{C}^c$

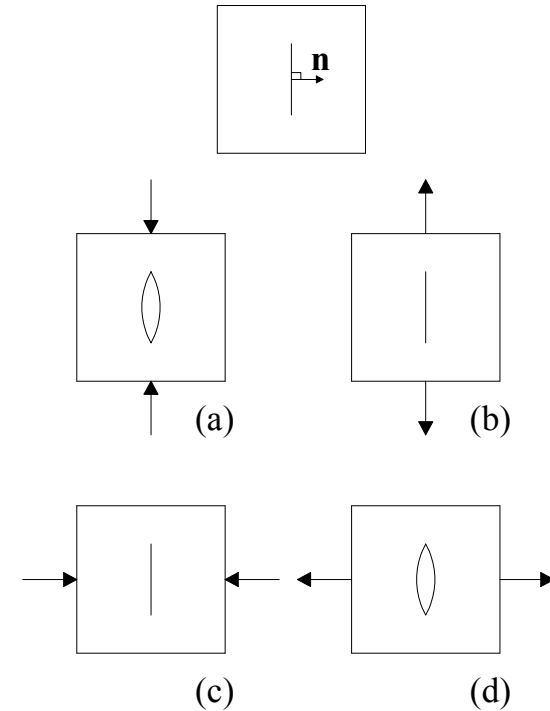
Opening and closing of microcracks:

$\mathbf{C}^c = \mathbf{P}^+ : \bar{\mathbf{C}}^c : \mathbf{P}^+$ (positive projections)

$\boldsymbol{\varepsilon} \approx \mathbf{C}^0 : \boldsymbol{\tau} + \mathbf{P}^+ (\bar{\mathbf{C}}^c : \boldsymbol{\tau}^+)$

$\nabla \boldsymbol{\varepsilon} \approx \mathbf{C}^0 : \nabla \boldsymbol{\tau} + \mathbf{P}^+ (\bar{\mathbf{C}}^c : (\nabla \boldsymbol{\tau})^+)$ Minimization problem solution for a given state of stress

$\bar{\mathbf{C}}^c = \bar{\mathbf{C}}_{IT}^c + \bar{\mathbf{C}}_{IC}^c$ Active microcracks: $\boldsymbol{\varepsilon}_{IT}^c \geq 0$ $\boldsymbol{\varepsilon}_{IC}^c \leq 0$



NON-LOCAL DAMAGE MODEL

Damage rules: $\dot{\bar{\mathbf{C}}}_{I_T}^c = \dot{\mu} \mathbf{R}_{I_T}(\boldsymbol{\tau})$ and $\dot{\bar{\mathbf{C}}}_{I_C}^c = \dot{\mu} \mathbf{R}_{I_C}(\boldsymbol{\tau})$

Irreversible character of damage: $\dot{\mu} \geq 0$

Internal length is assumed to be a function of damage: $g = g(\mu)$

Energy density dissipation:

$$d = \left(\begin{array}{l} \frac{1}{2} \boldsymbol{\tau}^+ : \mathbf{R}_{I_T} : \boldsymbol{\tau}^+ + \frac{1}{2} \boldsymbol{\tau}^- : \mathbf{R}_{I_C} : \boldsymbol{\tau}^- + \frac{1}{2} g^2 (\nabla \boldsymbol{\tau})^+ : \mathbf{R}_{I_T} : (\nabla \boldsymbol{\tau})^+ \\ + g \frac{dg}{d\mu} (\nabla \boldsymbol{\tau})^+ : \bar{\mathbf{C}}^c : (\nabla \boldsymbol{\tau})^+ \end{array} \right) \dot{\mu} - \dot{A}^c \geq 0$$

$\mathbf{R}_{I_T}, \mathbf{R}_{I_C}, \mathbf{C}_{I_T}^c, \mathbf{C}_{I_C}^c$ positive definite and $\dot{\mu} \geq 0 \Rightarrow dg / d\mu \geq 0$

Associated damage surface:

$$\Phi = \frac{1}{2} \boldsymbol{\tau}^+ : \boldsymbol{\tau}^+ + \frac{1}{2} c \boldsymbol{\tau}^- : \boldsymbol{\tau}^- - \frac{\pi}{2} t^2(\mu)$$

APPLICATION TO 4-POINT BENDING

4-point bending: an assumed stress-strain law is sufficient input information for damage characterization

Definition of damage

$$\tau_i = E_{0i} \varepsilon_i, \text{ for } \varepsilon_i \leq \varepsilon_{0i} \quad \tau_i = (1 - D_i) E_{0i} \varepsilon_i = \frac{E_{0i} \varepsilon_i}{1 + E_{0i} \mu_i}, \text{ for } \varepsilon_i > \varepsilon_{0i}$$

Assumed stress-strain law:

$$\tau_i = f_i \frac{\beta_i (\varepsilon / \varepsilon_i)}{\beta_i - 1 + (\varepsilon / \varepsilon_i)^{\beta_i}}$$

Damage functions:

$$D_i = 0 \text{ for } \varepsilon < \varepsilon_{0i} \quad D_i = 1 - \frac{\beta_i - 1 + (\varepsilon_{0i} / \varepsilon_i)^{\beta_i}}{\beta_i - 1 + (\varepsilon / \varepsilon_i)^{\beta_i}} \text{ for } \varepsilon \geq \varepsilon_{0i}$$

Young's modulus and isotropy:

$$E_{0i} = \frac{\beta_i f_i}{(\beta_i - 1 + (\varepsilon_{0i} / \varepsilon_i)^{\beta_i}) \varepsilon_i} \quad \frac{\varepsilon_t}{\varepsilon_c} = \frac{\beta_t f_t (\beta_c - 1 + (\varepsilon_{0c} / \varepsilon_c)^{\beta_c})}{\beta_c f_c (\beta_t - 1 + (\varepsilon_{0t} / \varepsilon_t)^{\beta_t})}$$

APPLICATION TO 4-POINT BENDING

Linear strain distribution: $\epsilon_{xx} = \epsilon_m + kz$

$$N = b \int_{-h/2}^{h/2} \sigma_{xx} dz = 0$$

Input: a given value for curvature, k

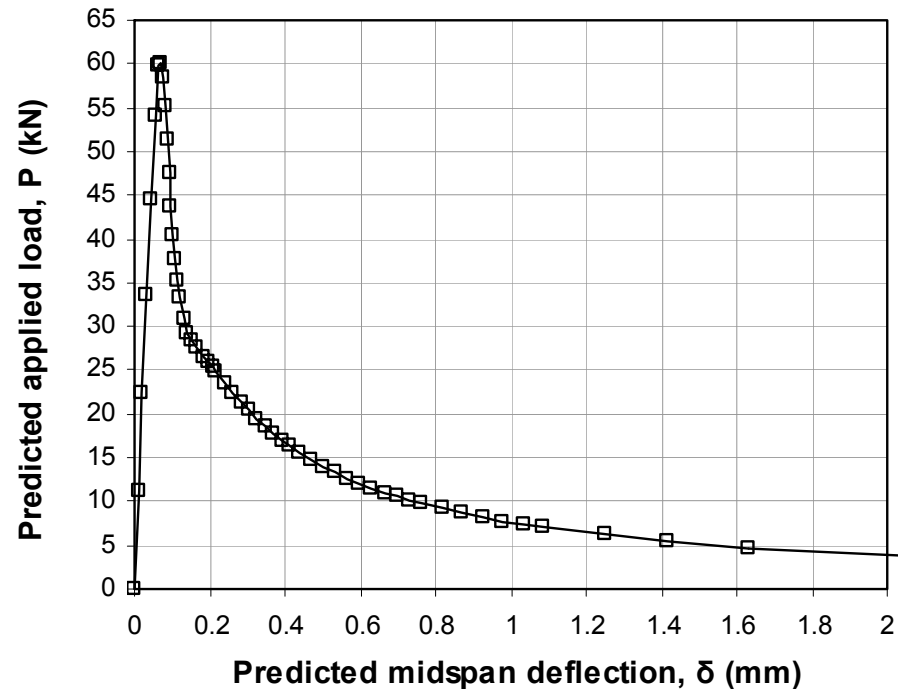
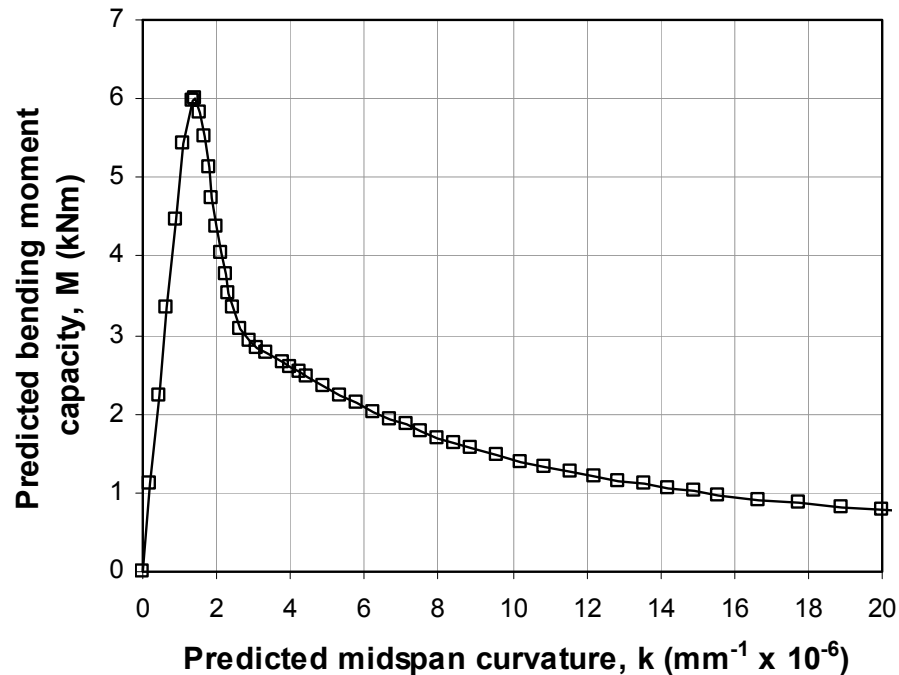
Find: ϵ_m that satisfies equilibrium (stress-strain law)

$$M = b \int_{-h/2}^{h/2} \sigma_{xx} z dz$$

Output: M , k point on the response diagram (size independent)

Input: k - δ kinematic relation (boundary value problem)

Final output: Force vs. midspan deflection diagram (size dependent)



APPLICATION TO 4-POINT BENDING

Non-local predictions:

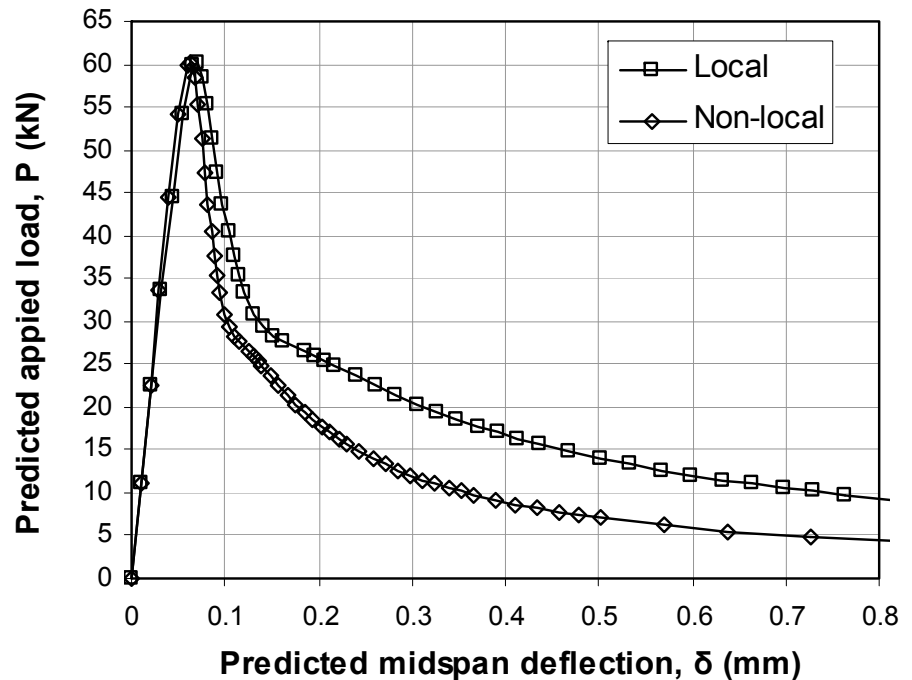
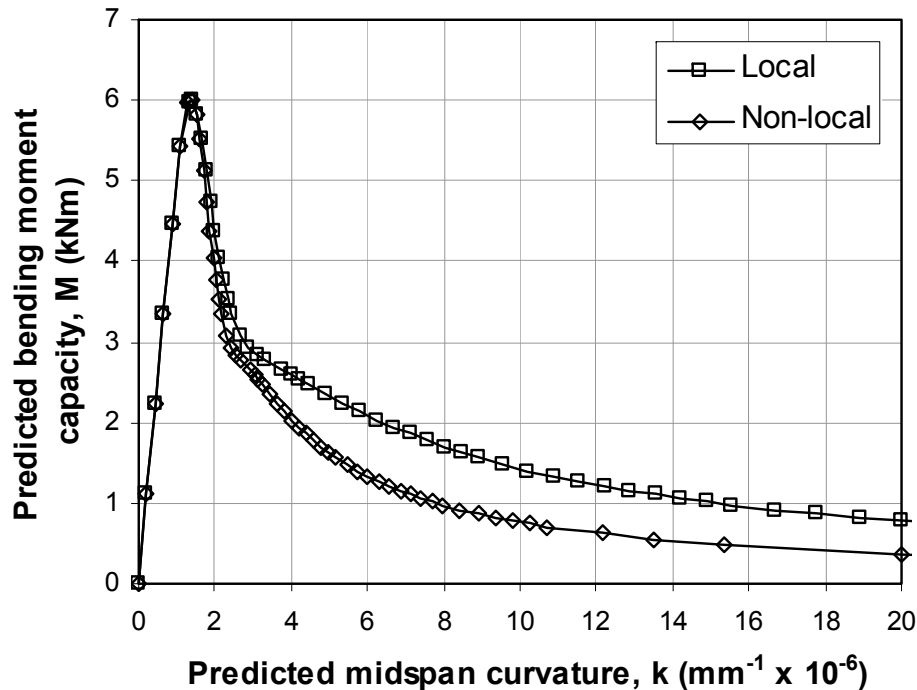
The local \mathbf{M} vs. \mathbf{k} is transformed to \mathbf{M} vs. \mathbf{k}_{grad} (scaling of curvature-elasticity)

The gradient solution of the boundary value problem is used to obtain the kinematic relation, $\delta_{\text{grad}}/\mathbf{k}_{\text{grad}} = \mathbf{f}(g/L)$

This relation is not constant but evolves with damage since $g=g(D)$

Internal length evolution law:

$$g = g_0 e^{nD}$$



APPLICATION TO 4-POINT BENDING

Objectivity of the model:

**Acoustic tensor
positive definite**

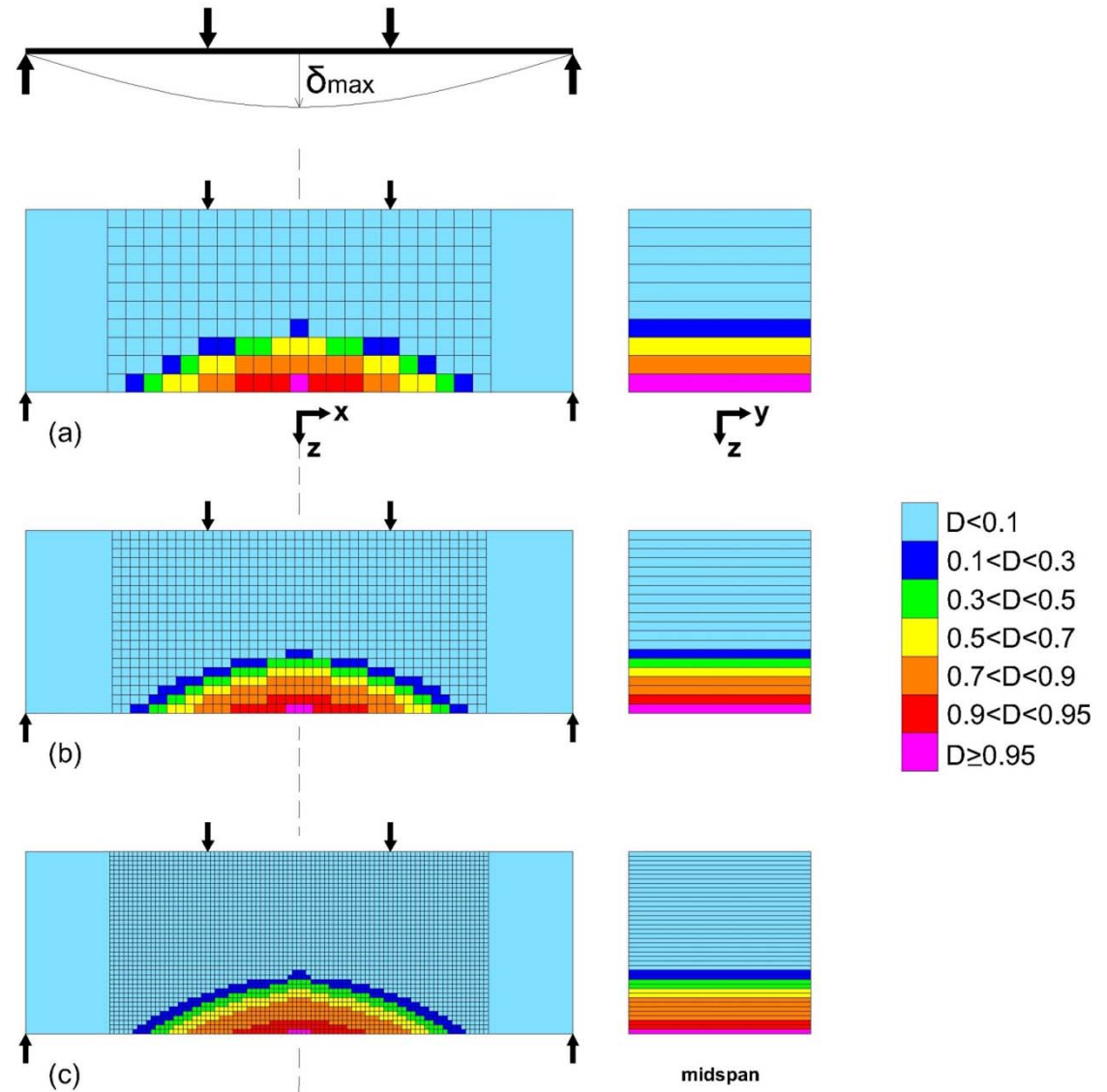
Good agreement with AE
findings:

- Initiation of softening ($D > 0$) and therefore microcracking

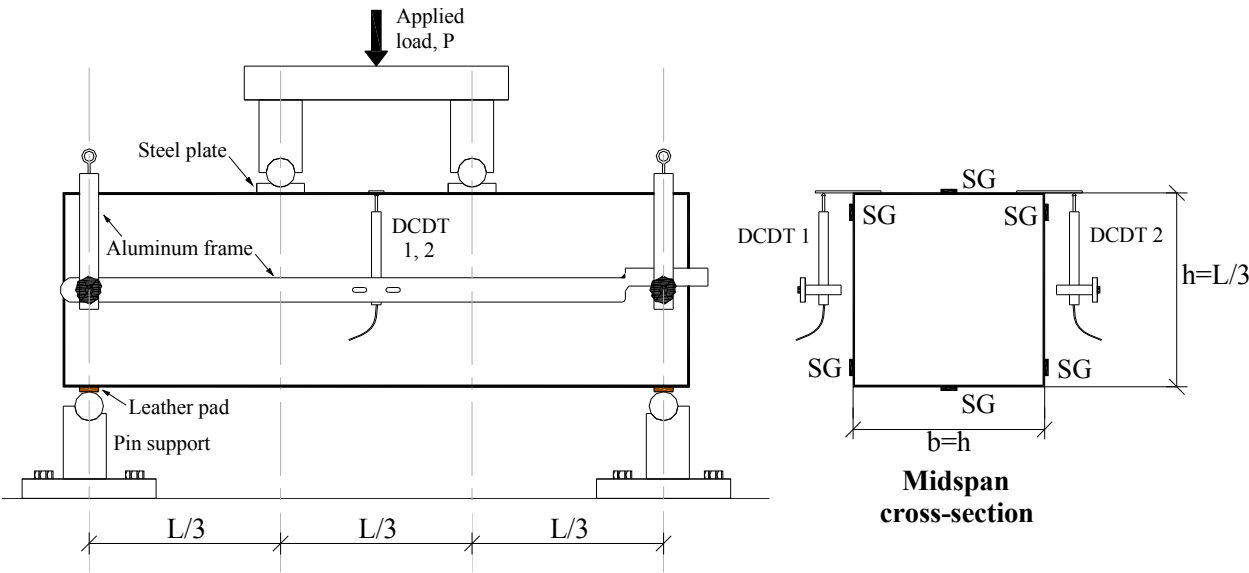
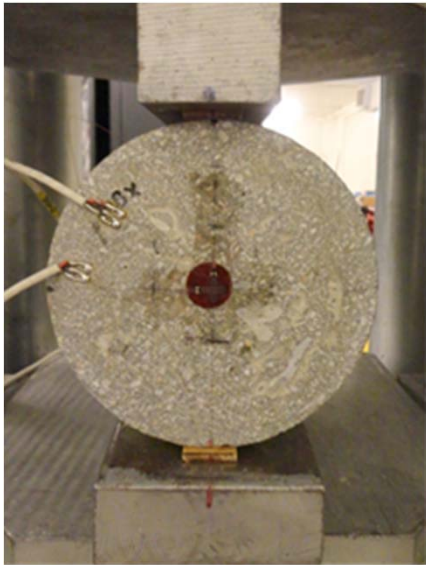
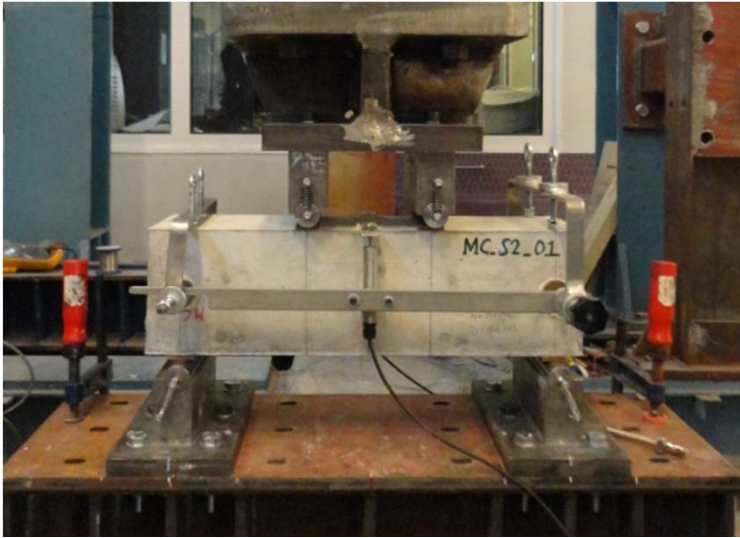
- Intense localization of AE energy release rate – macro-crack formation ($D > 0.95$)

Objective damage
characterization:

(for both $g=0$ and $g>0$)



EXPERIMENTAL PROGRAM



3D geometrical similar specimens

MATERIALS

Mix	Quantities (Kg/m ³)				Dry density (Kg/m ³)	Slump (cm)	Air-content ^(d) (%)
	Cement ^(a)	Aggregates ^(b)	Water (w/c ratio)	Additives ^(c)			
CM	450	1350	293 (0.65)	3.6	2100	- ^e	2.5
LC	208	1980	162 (0.78)	1.6	2335	25	3.0
NC	276	2080	176 (0.64)	1.5	2365	10	2.5
MC1	448	1720	204 (0.45)	4.0	2410	22.4	2.0
MC2	447	1640	207 (0.46)	6.0	2440	15.6	2.0

Sieve opening (mm)	% passing				
	LC	NC	MC1	MC2	CM
32	100	100	100	100	-
16	85.8	84.1	80.6	78.7	-
8	70.7	67.8	60.0	57.7	-
4	62.7	59.7	49.6	49.1	-
2	45.4	43.3	35.7	35.5	-
1	29.6	28.2	23.3	23.2	100
0.5	-	-	-	-	30
0.25	12.9	12.3	10.2	10.1	-
0.075	8.0	7.6	6.3	6.3	-

Four concrete mixes $d_{max}=32$ mm

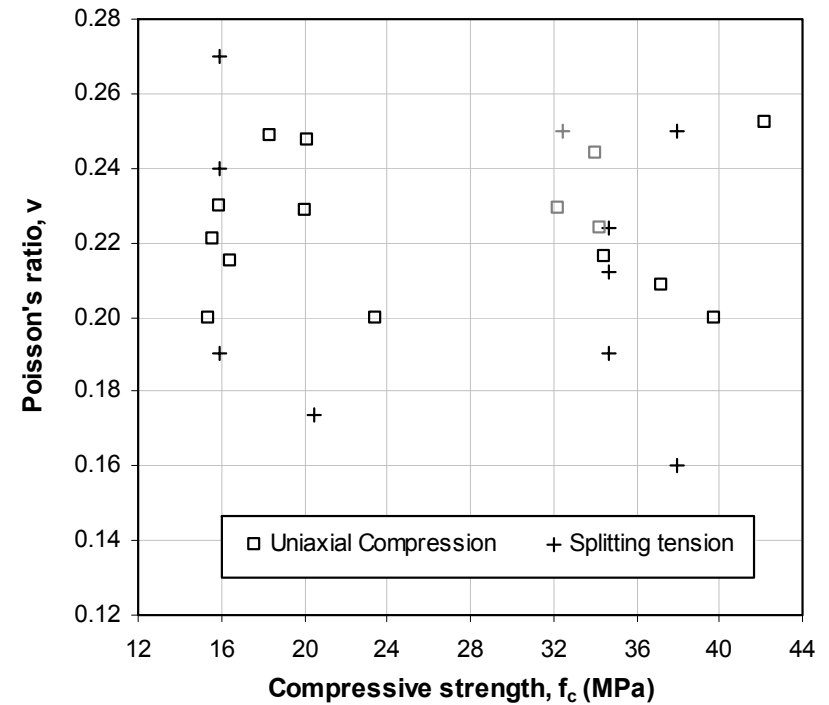
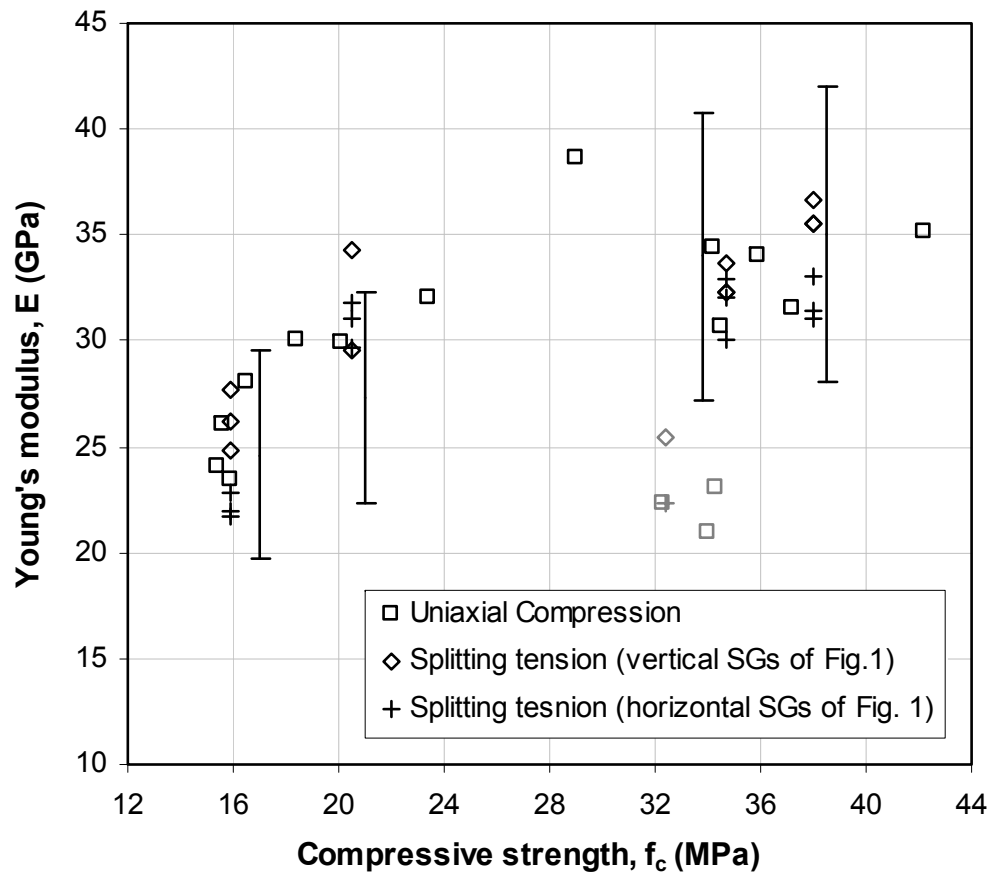
Low-strength (LC)

Normal-strength (NC)

Medium-strength (MC1 and MC2)

Cement mortar $d_{max}=1$ mm

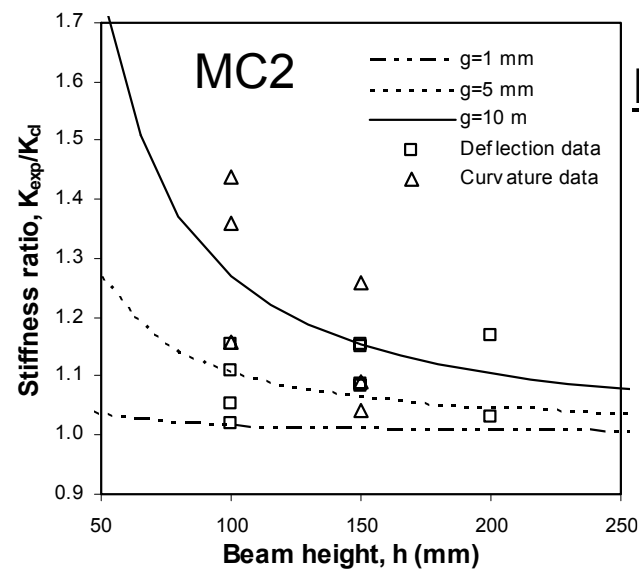
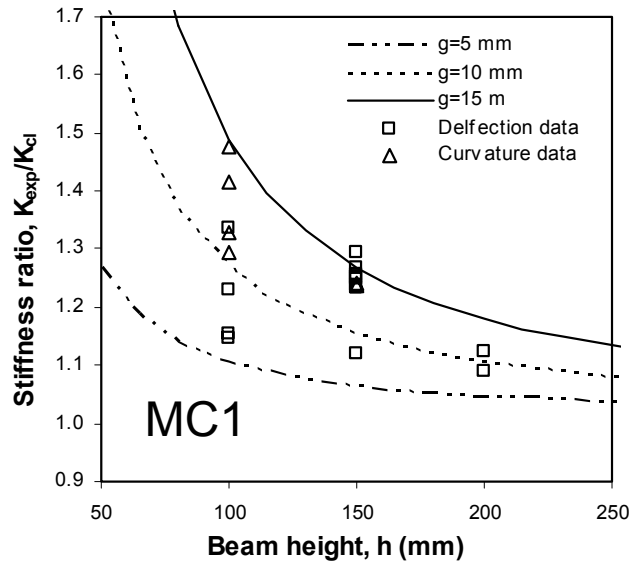
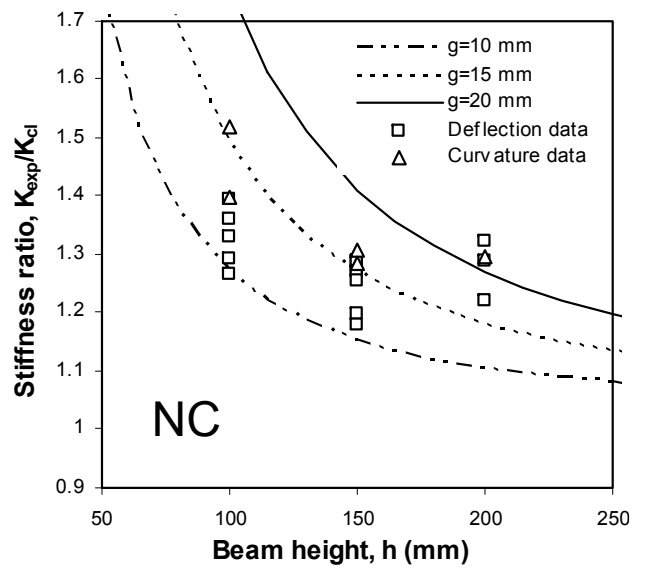
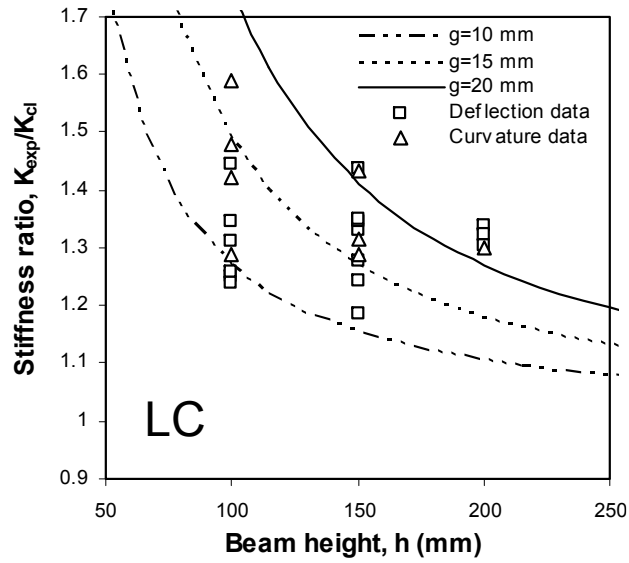
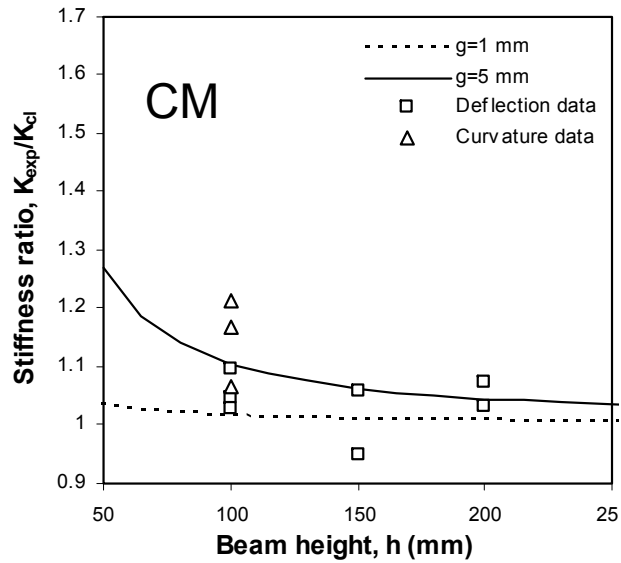
CLASSICAL ELASTICITY PROPERTIES



Classical elasticity properties used in the analysis

Mix	LC	NC	MC1	MC2	CM
E, GPa	25.0	30.70	32.7	34.0	22.3
ν	0.2				

INTERNAL LENGTH AND SIZE EFFECT IN ELASTICITY



Elasticity: $P=K \delta$ & $P=S k$

Size effect in elasticity

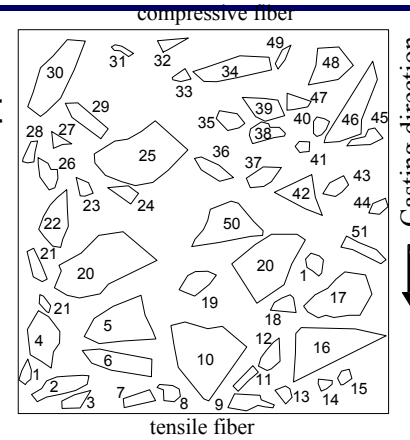
if: $K_{exp}/K_{cl} > 1$

$S_{exp}/S_{cl} > 1$

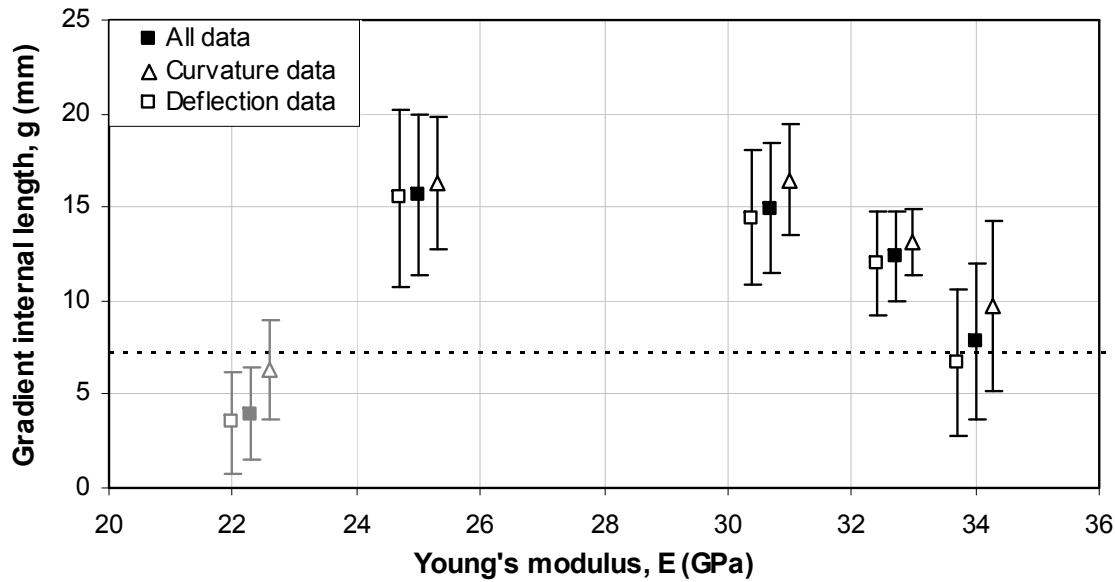
INTERNAL LENGTH AND MICROSTRUCTURE



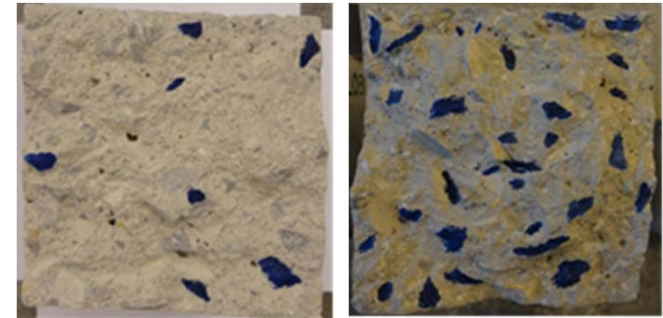
Concrete contains different aggregate sizes in different volume fractions



Average inclusion size ranges between **10 to 20 mm**

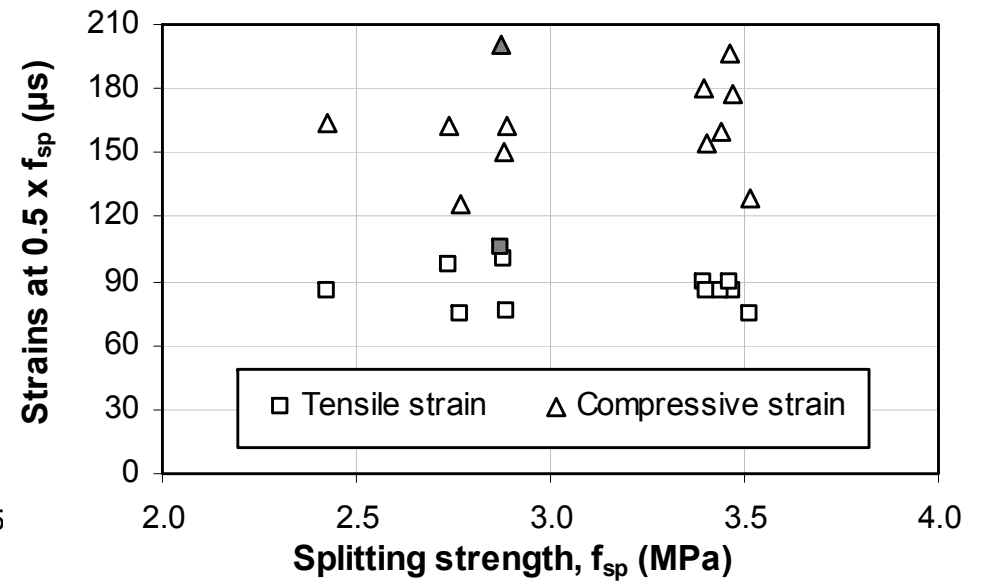
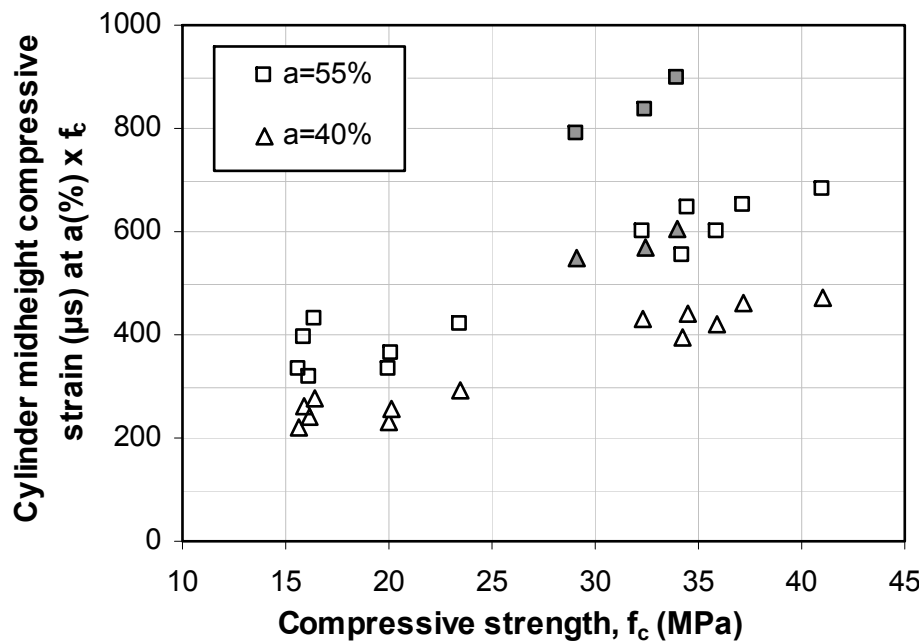


Fracture surfaces and fractured aggregates

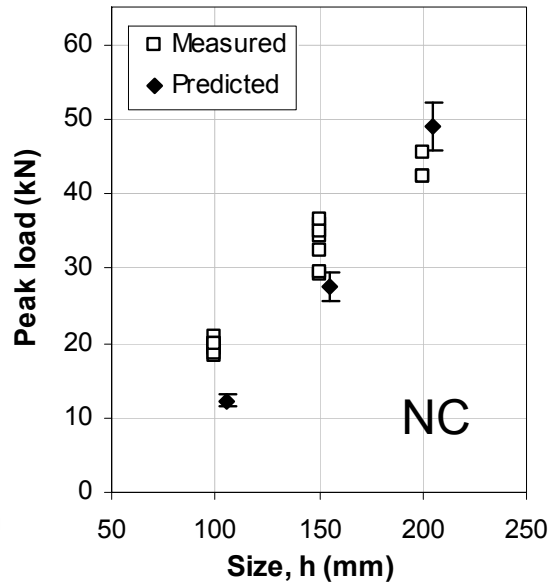
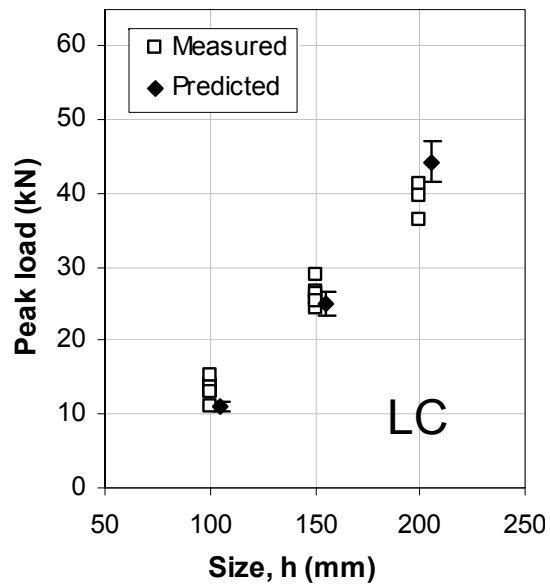


INELASTICITY: PEAK LOAD

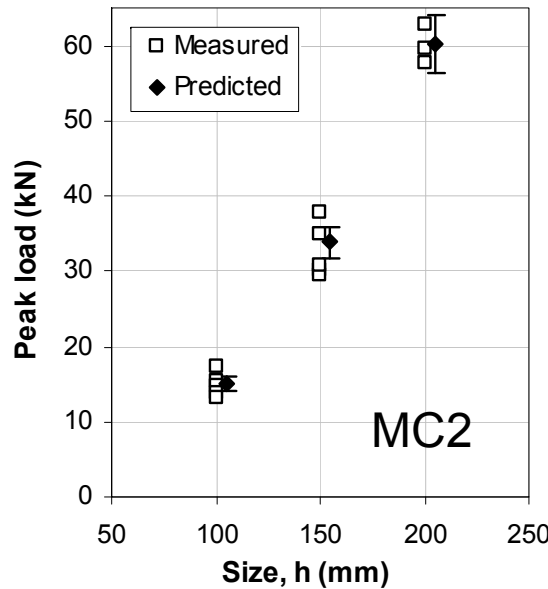
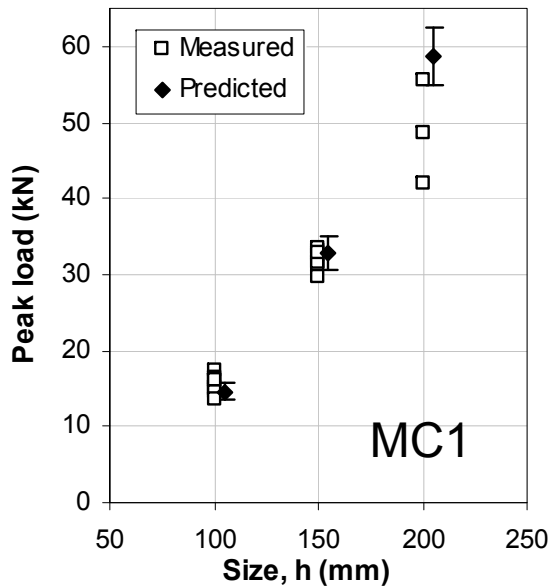
Mix	Uniaxial stress-strain law parameters								Young's modulus
	Compression				Tension				$E^{(a)}$ (GPa)
	$f_c^{(a)}$	ϵ_c	$\epsilon_{0c}^{(b)}$	$b_c^{(b)}$	$f_t^{(c)}/f_{sp}^{(a)}$	$b_t^{(c)}$	$\epsilon_{0t}^{(d)}$	$\epsilon_t^{(d)}$	
	(MPa)	($\times 10^{-6}$)	($\times 10^{-6}$)	-	-	-	($\times 10^{-6}$)	($\times 10^{-6}$)	
LC	15.9	1500	250	1.643	0.80	4.5	70	110	25.0
NC	20.5	1500	270	1.707	0.85	5.0	65	100	30.7
MC1	34.7	1500	430	3.360	0.88	6.0	80	120	32.7
MC2	38.0	1500	450	3.890	0.90	6.5	80	110	34.0
CM ^(e)	32.4	1800	600	5.183	0.95	10	130	130	22.4



INELASTICITY: PEAK LOAD



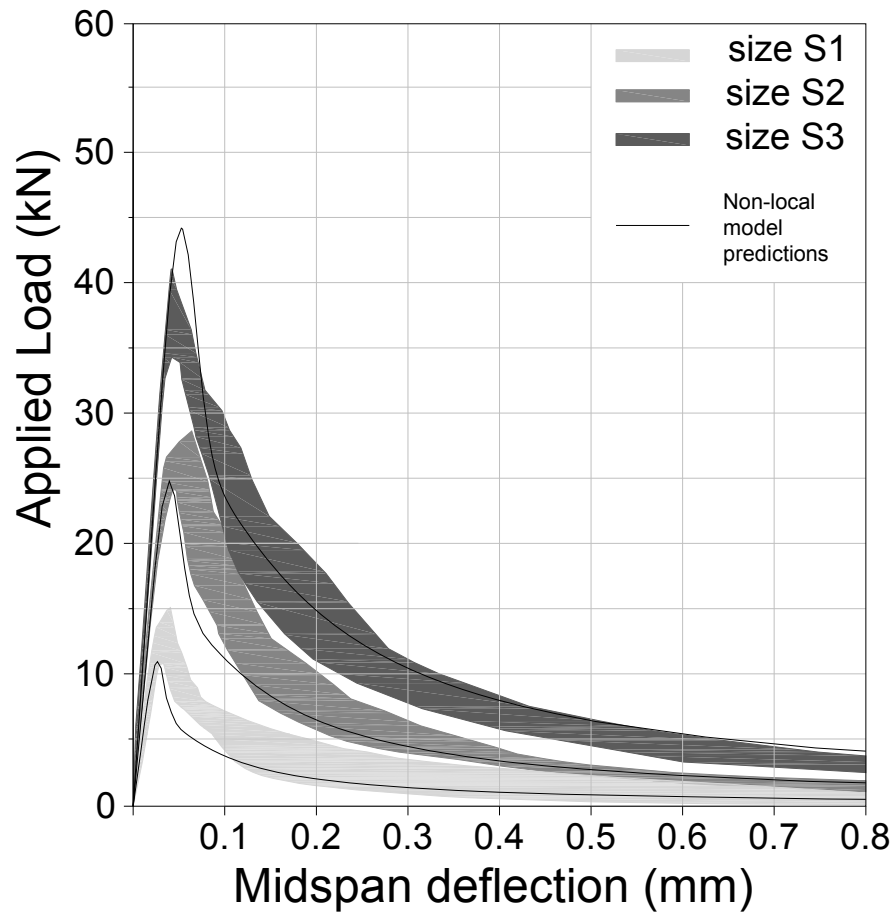
Model predictions correspond to a size independent flexural strength



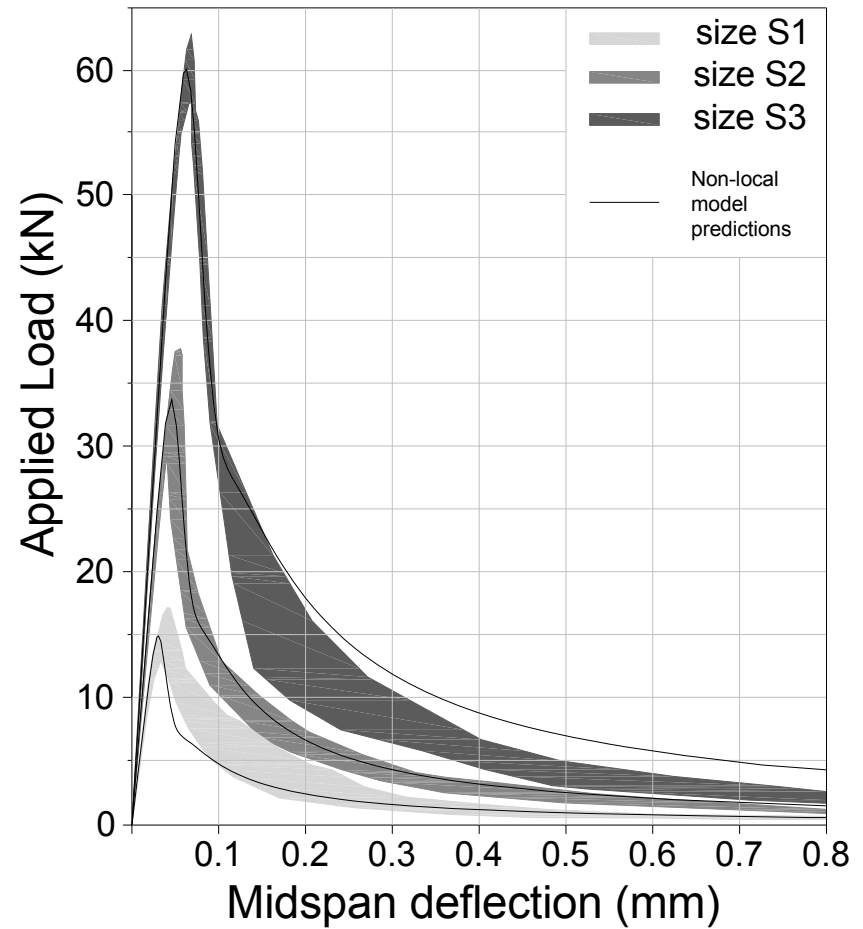
Scatter of predicted values corresponds to a +/- 5% deviation of the assumed material's tensile strength

INELASTICITY: SOFTENING BRANCH

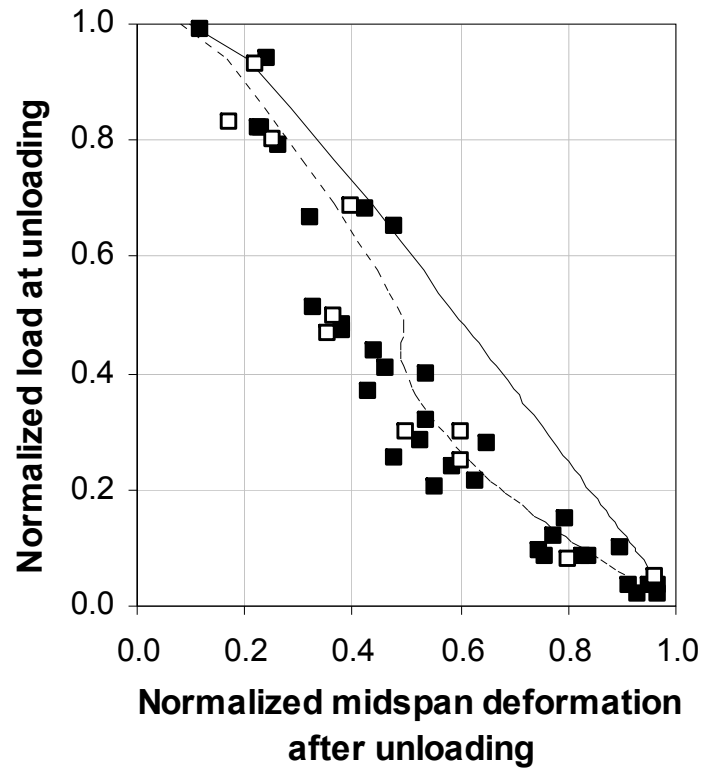
LC mix: $g = 17e^{0.9D}$ (mm)



MC2 mix: $g = 8e^{2D}$ (mm)

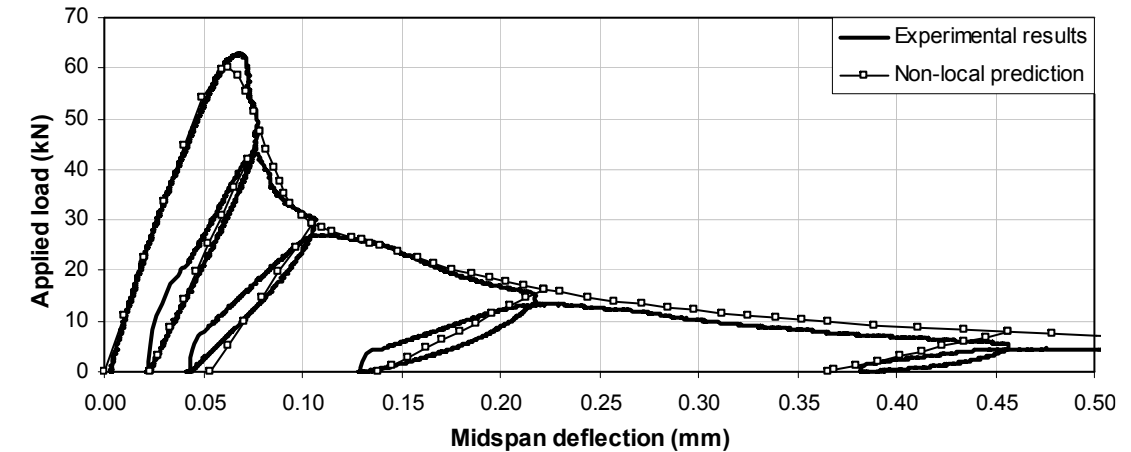
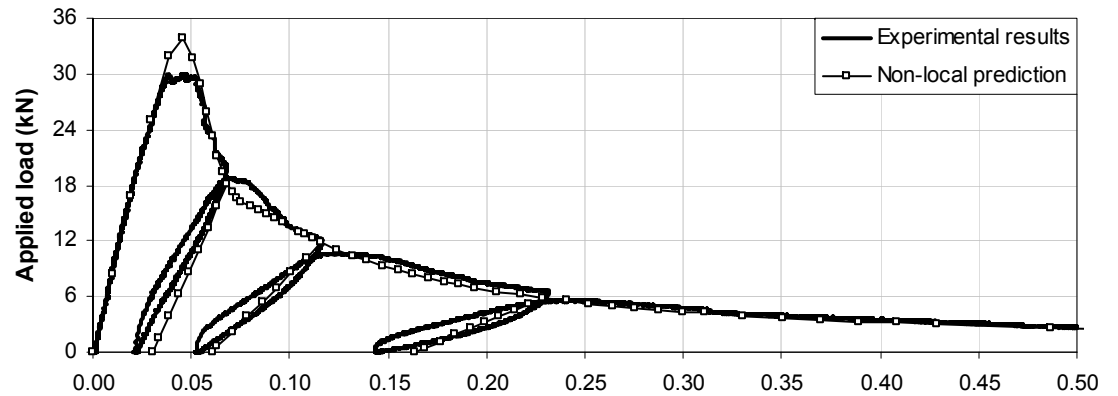
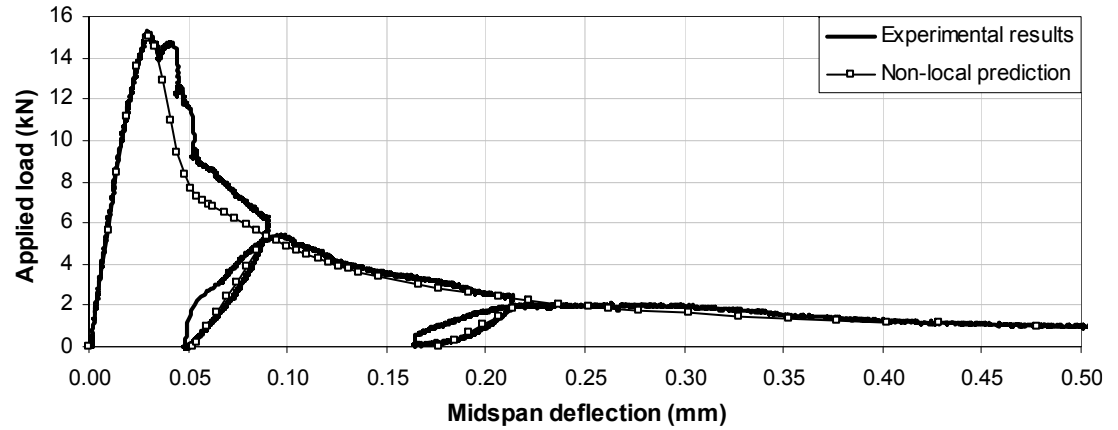


INELASTICITY: UNLOADING PATH



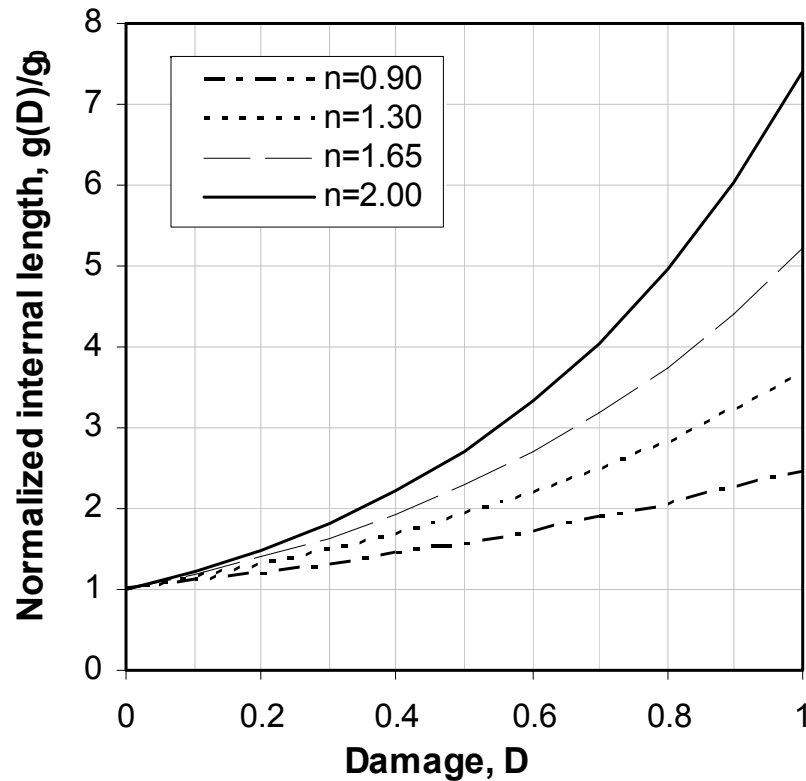
$$\delta_{pl} = \bar{\delta} - \bar{P} / (1 - \bar{D}) K_0$$

MC2 mix



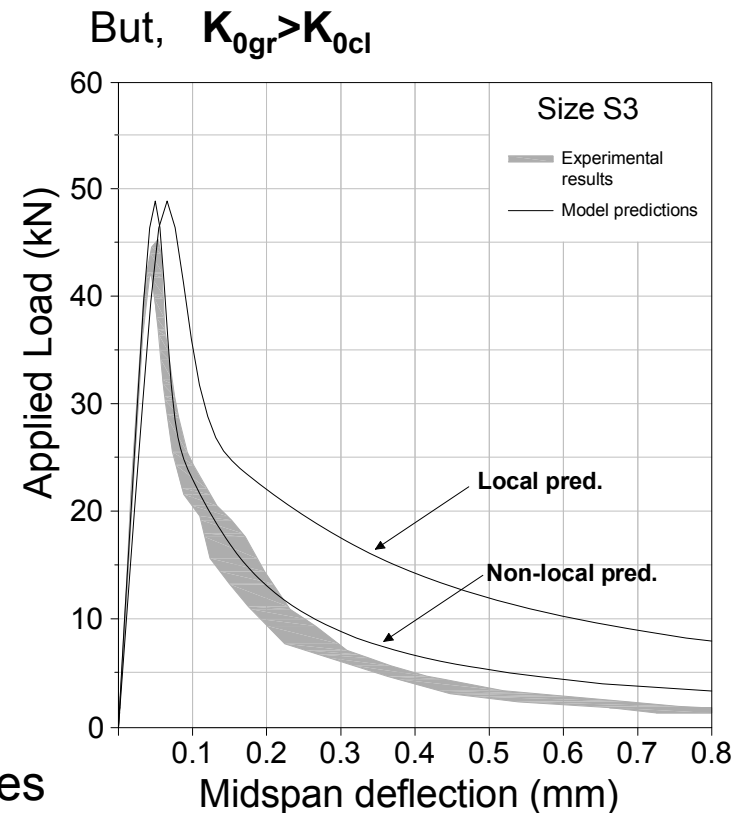
EVOLUTION LAW OF GRADIENT INTERNAL LENGTH

Non-local parameters	Mix			
	LC	NC	MC1	MC2
g_0 (mm)	17	15	12.5	8.0
n	0.90	1.30	1.65	2.00

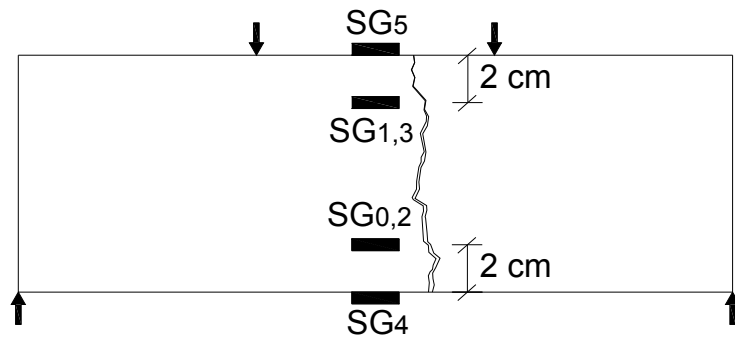


n increases as brittleness increases in the mixes

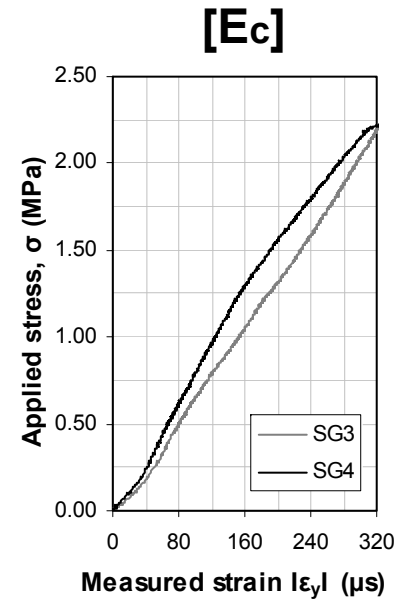
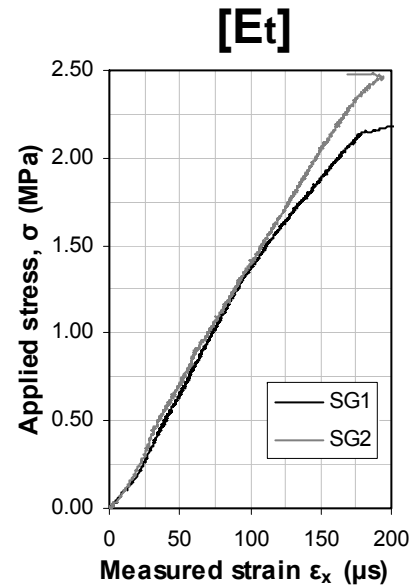
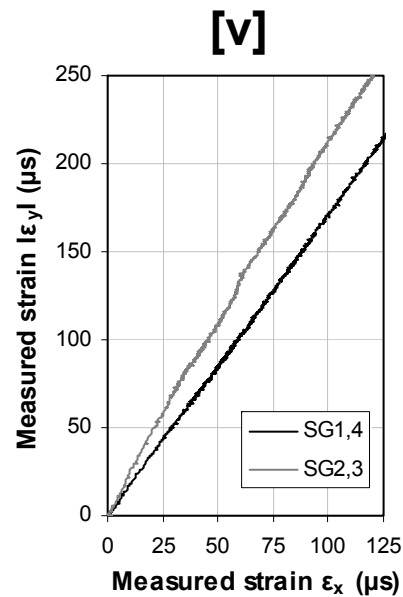
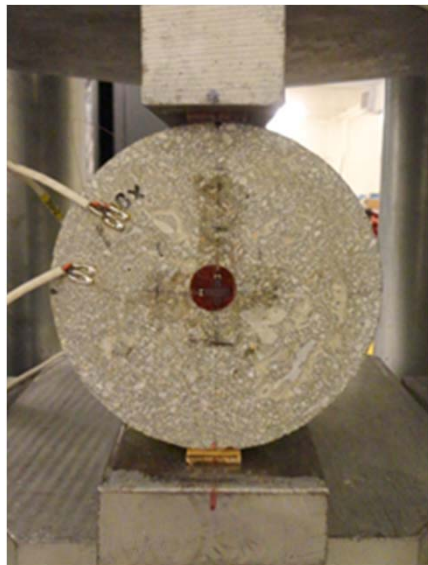
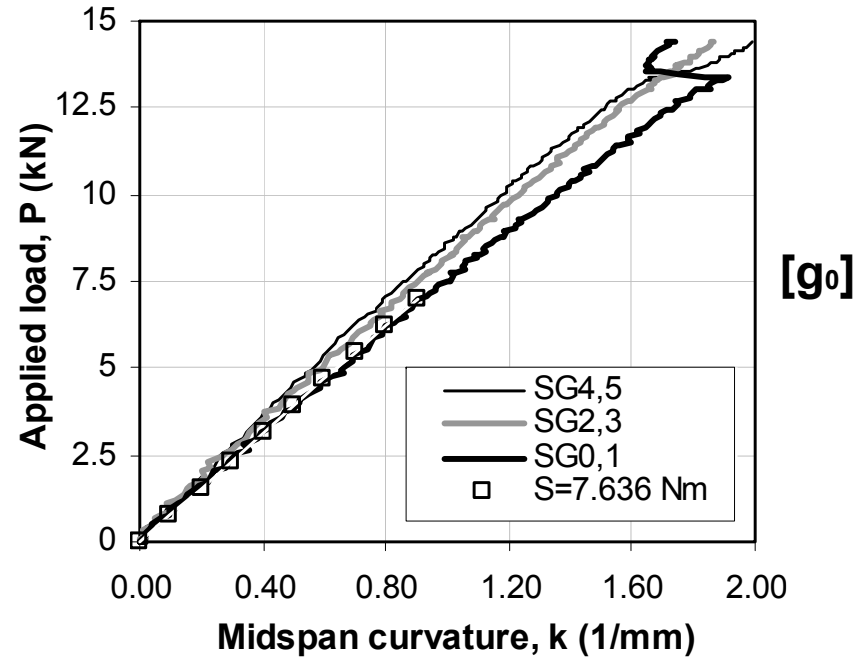
- Progressively stiffer response if $dg/dD > 0$ (size effect in inelasticity)
- Cauchy stiffness decreases with damage: $dK/dD = -K_0 < 0$



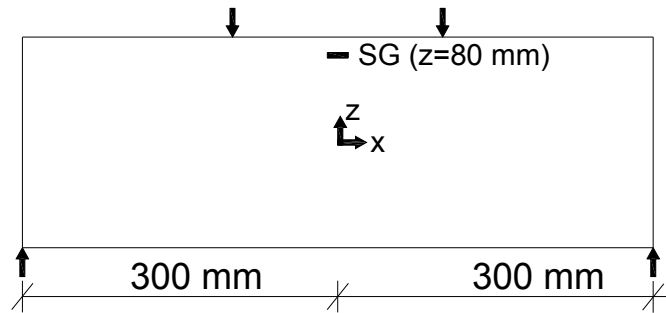
STRAIN GAGE MEASUREMENTS



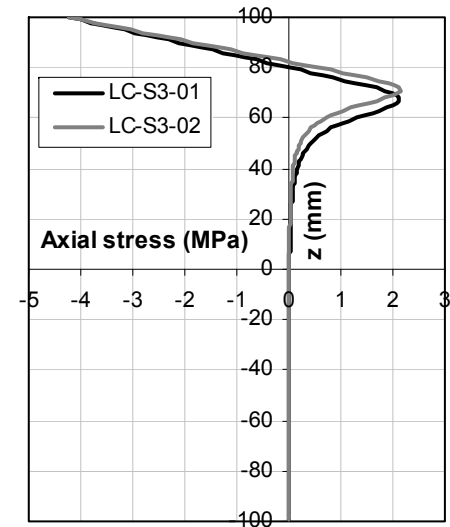
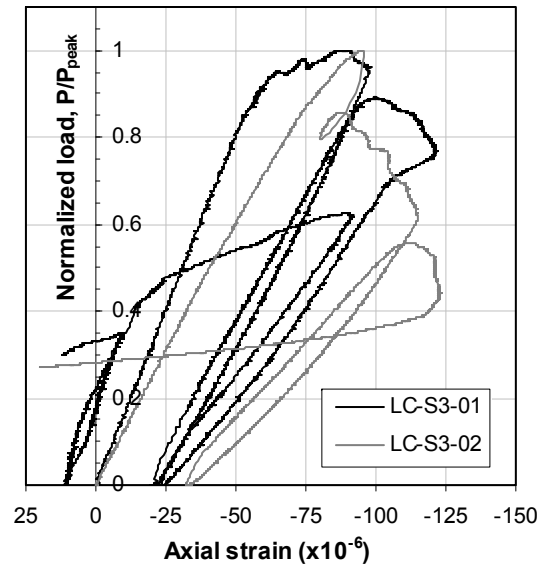
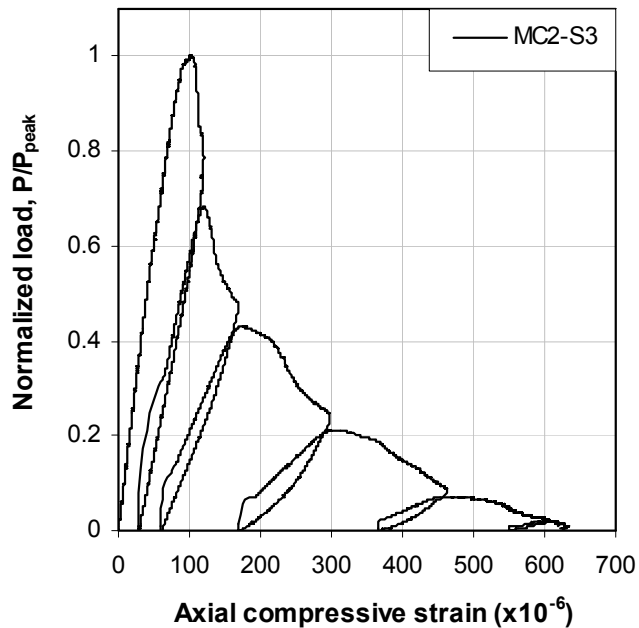
For $g > 0$, $S_{exp} > S_{cl}$ or $k_{exp} < k_{cl}$
 (P=Sk)



STRAIN GAGE MEASUREMENTS



Damage characterization



Plastic strains: verification of progressively stiffer response in inelasticity

$$\sigma(\varepsilon, \varepsilon_{,xx}) = (1 - D(\varepsilon))E(\varepsilon - g^2 \varepsilon_{,xx})$$

SIZE EFFECTS

Size effect in elasticity ($g > 0$)

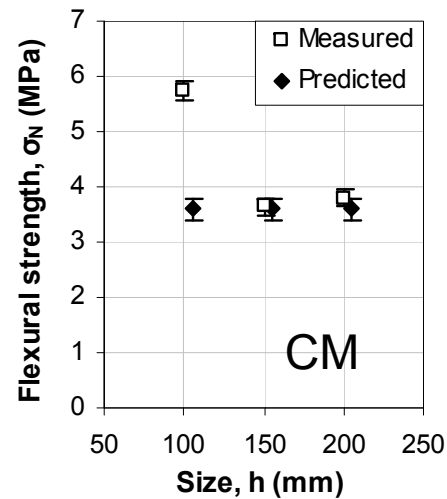
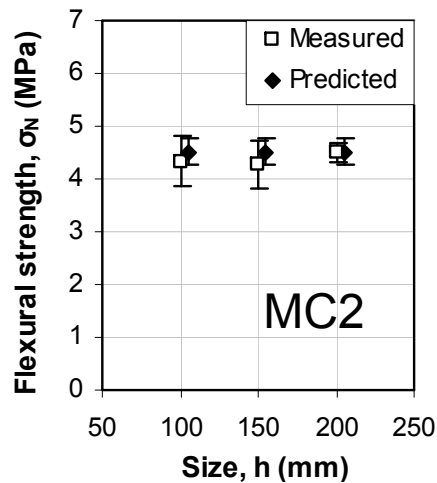
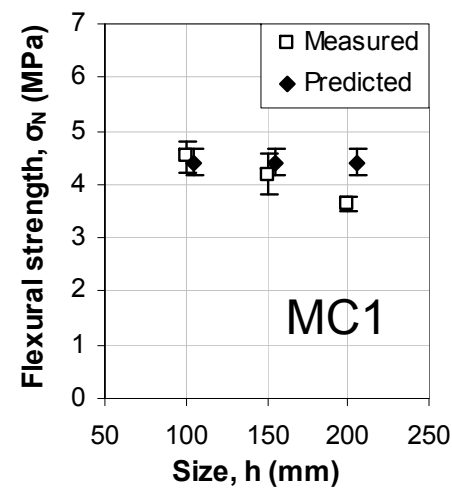
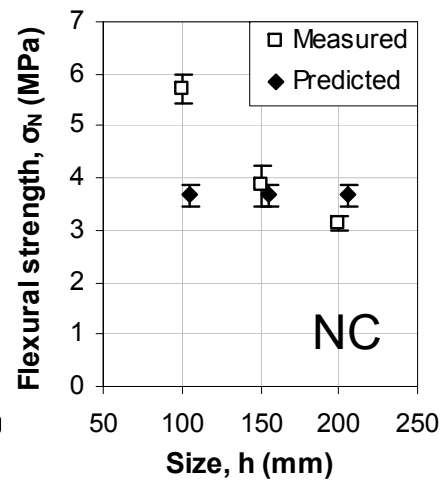
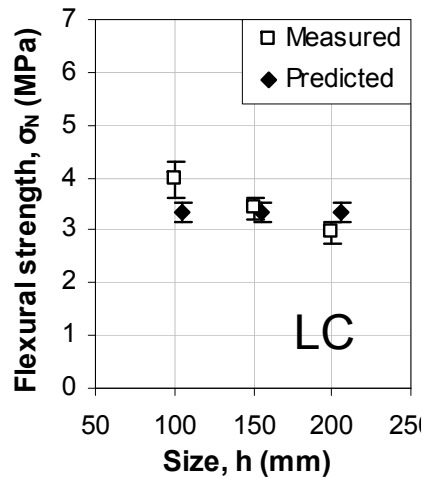
Size effect in inelasticity ($g = g(D)$ with $dg/dD > 0$)

Size effect on strength

Sources/Explanations of size effect on strength

1. Fracture mechanics size effect
 2. Statistical
 3. Stress redistribution at meso-scale (Lattice models)
 4. Multi-fractal size effect
1. Diffusion phenomena
 2. Hydration heat – microcracking during cooling
 3. Wall effect due to inhomogeneous boundary layer

SIZE EFFECT ON FLEXURAL STRENGTH

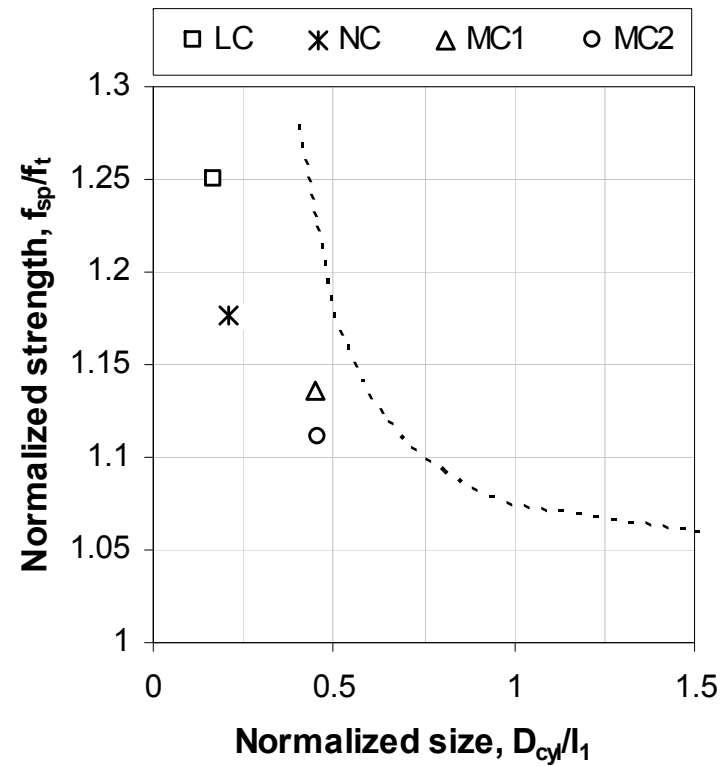
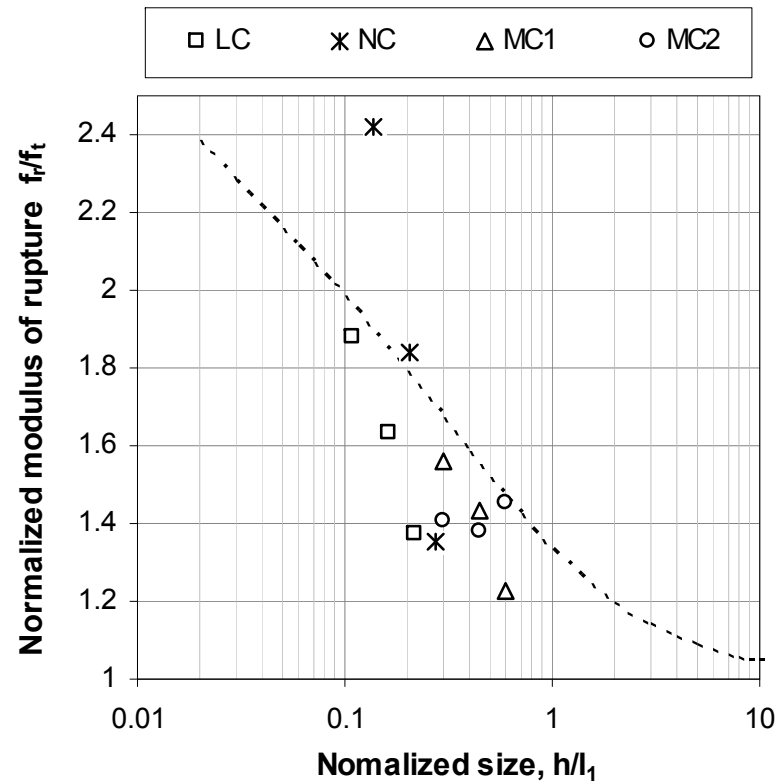


Statistical size effect

$$\sigma_N \propto h^{-1/8}$$

Observed size effect on flexural strength cannot be attributed to statistical sources

SIZE EFFECT ON FLEXURAL STRENGTH



Limited scale range , **1 : 1.5 : 2**,
for examining size effect on strength

CONCLUDING REMARKS

1. Experimental quantification of the internal length assumed by simplified dipolar elasticity

3. Physical correlation between the internal length and material's microstructure

5. Use of SG's and continuous strain gradient models

7. Size effect on strength

2. Concrete can be seen as a model material for studying size effects in elasticity

4. Experimental verification of key assumptions

6. Further experimental work is needed

FUTURE WORK

Introducing size effect on strength in the proposed model for the case of concrete

- Refinement of the model
 Include effect of distributed damage to the Timoshenko model
- Size dependent damage characterization (Van Vliet & Van Mier)
- Damage characterization not based on Cauchy strains alone: $D(\varepsilon, \varepsilon_{,x})$
- Strain gradient lattice models predicting size effect on strength ?

FUTURE WORK

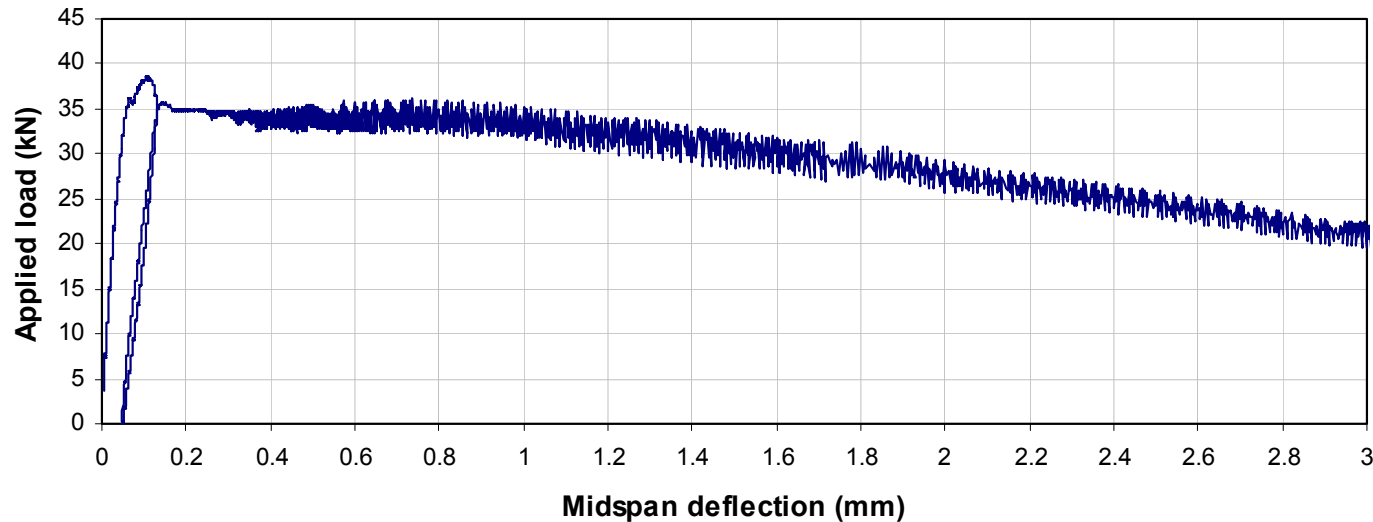
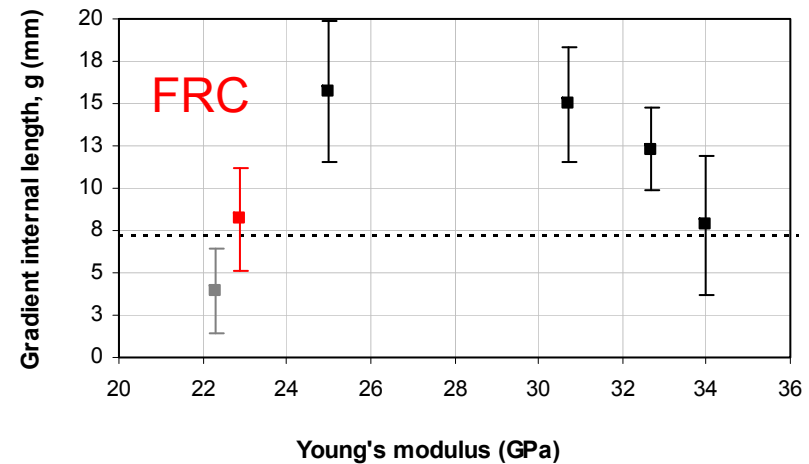
Fiber Reinforced Concrete (FRC)

4-point bending experiments

g_0 of FRC = 8 mm

g_0 of matrix = 4 mm

same order of magnitude as predicted by the simplified 2D case



Very ductile softening response – b_t parameter of stress-strain in the range of 1.5

For FRC, $n \rightarrow 0$. Therefore, $g(D) = g_0$ (the thermodynamic limit of $dg/dD = 0$ is recovered)

FUTURE WORK

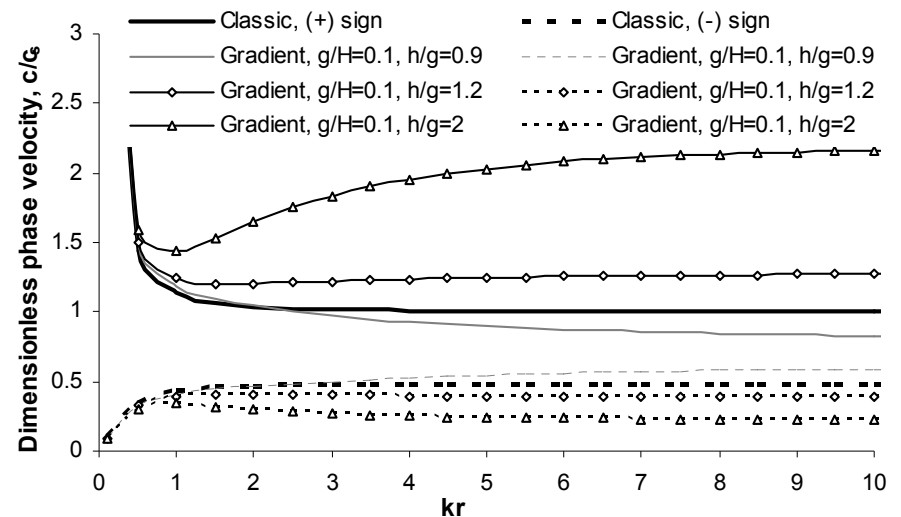
Micro-inertia length and dynamic loading:

Hamilton's principle:
$$\delta \int_{t_0}^{t_1} (U_{\text{tot}} - W - K) dt = 0$$

2 Eqs. of motion and 2 gradient length: stiffness **related g** and inertia **related h**

$$\left(1 + \frac{A}{I} g^2 - g^2 \frac{d^2}{dx^2}\right) \frac{d\bar{M}}{dx} - \left(1 - g^2 \frac{d^2}{dx^2}\right) \bar{Q} = \left(1 + h^2 \frac{A}{I} - h^2 \frac{d^2}{dx^2}\right) m_I \ddot{\psi}$$

$$\left(1 - g^2 \frac{d^2}{dx^2}\right) \frac{d\bar{Q}}{dx} = -q + \left(1 - h^2 \frac{d^2}{dx^2}\right) m_A \ddot{w}$$



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(Director: Prof. A. E. Giannakopoulos)



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PhD presentation on

**“Size Effects in Semi-brittle Materials and
Gradient Theories with Application to Concrete”**

By

Antonios Triantafyllou

Department of Civil Engineering
School of Engineering, University of Thessaly
Volos, Greece



Advisor: Prof. Philip C. Perdikaris

Co-advisor: Prof. Antonios E. Giannakopoulos